

Test in English for Maintenance Licenses

(Last Update: January 2015)

Subject : ENGLISH FOR MAINTENANCE LICENSES.

Number of Questions : 300

- 1.- WHAT DOES THE AIR FLOW PRODUCE OVER ANY AERODYNAMIC SURFACE? (19030) REF.: AC 65-15A, PAGE 64.
 - A.- PRODUCES AN INCREASE IN VELOCITY AND TEMPERATURE.
 - B.- PRODUCES A REDUCTION IN VELOCITY AND TEMPERATURE.
 - C.- PRODUCES CERTAIN REDUCTION IN VELOCITY WITH CORRESPONDING INCREASES IN TEMPERATURE.
 - D.- PRODUCES CERTAIN INCREASES IN VELOCITY WITH CORRESPONDING REDUCTION IN TEMPERATURE.
- 2.- HOW MANY GENERAL TYPES OF EXHAUST SYSTEMS IN USE ON RECIPROCATING AIRCRAFT ENGINES ARE THERE? (19299) REF.: AC 12A, PAGE 96.
 - A.- FOUR.
 - B.- TWO.
 - C.- ONLY ONE.
 - D.- THE TYPES OF EXHAUST SYSTEMS DEPEND OF ENGINE MANUFACTURER.

3.- IN MOST CASES, WHAT LOADS ARE DESIRABLE FOR STRUCTURAL MEMBERS TO CARRY ON? (18839) REF.: AC 65-15A, PAGE 25.

- A.- END LOADS RATHER THAN SIDE LOADS.
- B.- SIDE LOADS RATHER THAN END LOADS.
- C.- AERODYNAMIC LOADS RATHER THAN STRUCTURAL LOADS.
- D.- STRESS LOADS RATHER THAN STRENGTH LOADS.

4.- WHERE ARE MOST FLAPS HINGED? (19284) REF.: AC 65-15A, PAGE 21.

- A.- OVER SPECIAL PYLON.
- B.- TO THE MIDDLE SECTION OF THE WINGS.
- C.- TO THE LOWER TRAILING EDGES OF THE WINGS.
- D.- TO THE UPPER TRAILING EDGES OF THE WINGS.

5.- WHICH ARE THE MOST COMMON SYNTHETIC FIBERS USED IN AIRCRAFT? (18843) REF.: AC 65-15A, PAGE 85.

- A.- NYLON, ORLON AND DACRON.
- B.- NYLON, COTTON AND LINEN.
- C.- FABRIC, WARP AND PLY.
- D.- WOOF, WOVEN AND FILLING.

6.- WHAT IS PROVIDED AT MANY PITOT-STATIC TUBES IN ORDER TO PREVENT ICING DURING FLIGHT? (19273) REF.: AC 65-15A, PAGE 475.

- A.- CHEMICAL HEATING ELEMENTS.
- B.- AERODYNAMIC HEATING ELEMENTS.
- C.- HOT AIR HEATING ELEMENTS.
- D.- ELECTRICAL HEATING ELEMENTS.

- 7.- HOW MANY FACTORS DETERMINE THE AMOUNT OF LIFT AVAILABLE IN HELICOPTER OPERATION? (19025) REF.: AC 65-15A, PAGE 56.
 - A.- FEW FACTORS.
 - B.- ONE FACTOR.
 - C.- MANY FACTORS.
 - D.- SOME FACTORS.
- 8.- WHAT WILL THE AVIATION MECHANIC NEED THE MATHEMATICS TOOLS FOR? (18646) REF.: FAA-H-8083-30, PAGE 1-1.
 - A.- TO SEE, WASH, AND DISPATCH THE AIRPLANE.
 - B.- TO WORK, TEST AND CHECK THE AIRPLANES.
 - C.- TO REPAIR, MAINTAIN AND CERTIFY AIRPLANES.
 - D.- TO CONTROL, DISPATCH AND PARK THE AIRPLANES.
- 9.- WHAT OLD MATERIAL IS STILL IN USE TODAY TO COVER DIFFERENT AIRCRAFT PARTS? (18844) REF.: AC 65-15A, PAGE 85.
 - A.- WOOD.
 - B.- FABRIC.
 - C.- PLASTIC.
 - D.- COMPOSITE.
- 10.- WHAT LANDING GEAR ARRANGEMENT IS THE MOST USED IN MODERN AIRCRAFT? (19269) REF.: AC 65-15A, PAGE 341.
 - A.- A TAIL WHEEL AND A NOSE SKID GEAR ARRANGEMENT.
 - B.- A TAIL SKID ARRANGEMENT.
 - C.- A TAIL WHEEL GEAR ARRANGEMENT.
 - D.- A TRICYCLE GEAR ARRANGEMENT.
- 11.- WHAT ARE THE TYPES OF MAGNETS? (19296) REF.: FAA-H-8083-30, PAGE 10-11.
 - A.- BIG AND SMALL OR LEFT AND RIGHT.
 - B.- NATURAL OR ELECTRICAL.
 - C.- NATURAL OR ARTIFICIAL.
 - D.- NATURAL AND ARTIFICIAL.

12.- WHAT IS THE DEFINITION OF MAGNETISM? (19293) REF.: FAA-H-8083-30, PAGE 10-7.

- A.- IS DEFINED AS THE QUALITY OF AN OBJECT TO ATTRACT THE IRON METAL.
- B.- IS THE PROPERTY OF AN OBJECT TO ATTRACT ALL SUBSTANCES.
- C.- IS THE PROPERTY OF AN OBJECT TO ATTRACT ALL METALLIC SUBSTANCES.
- D.- IS THE PROPERTY OF AN OBJECT TO ATTRACT CERTAIN METALLIC SUBSTANCES.

13.- USUALLY, WHERE DOES THE MAGNESIUM SKIN CORROSION OCCUR? (18787) REF.: FAA-H-8083-30, PAGE 6-15.

- A.- AROUND THE INSULATING WASHERS.
- B.- OVER WASHERS AND BOLTS.
- C.- NEVER AROUND EDGES OF SKIN PANELS.
- D.- AROUND EDGES OF SKIN PANELS.

14.- WHY CAN BE CONSTRUCTED THE FUSELAGES IN FIXED-WING AIRCRAFT AND HELICOPTERS? (18822) REF.: AC 65-15A, PAGE 24.

- A.- FUSELAGES MAY BE CONSTRUCTED WELDED, BOLTED TRUSS OR SOME FORM OF SEMI-MONOCOQUE CONSTRUCTION.
- B.- FUSELAGES MAY BE WELDED TRUSS OR SOME FORM OF MONOCOQUE CONSTRUCTION.
- C.- FUSELAGES MAY BE RIVETED, WELDED, GLUED OR SOME FORM OF MONOCOQUE CONSTRUCTION.
- D.- FUSELAGES ARE CONSTRUCTED ONLY LIKE SEMI-MONOCOQUE OR MONOCOQUE CONSTRUCTION.

15.- WHAT IS LIFT FORCE IN A HELICOPTER? (18980) REF.: AC 65-15A, PAGE 50.

- A.- IS THE FORCE REQUIRED TO FLY THE HELICOPTER.
- B.- IS THE FORCE REQUIRED TO SUPPORT THE WEIGHT OF THE HELICOPTER.
- C.- IS THE FORCE REQUIRED TO SUPPORT THE TORQUE ROTORS.
- D.- IS THE FORCE REQUIRED TO LOAD AND UNLOAD THE HELICOPTER.

- 16.- KEEPING THE INTERIOR OF THE AIRCRAFT CLEAN IS JUST AS IMPORTANT AS MAINTAINING? (18793) REF.: FAA-H-8083-30, PAGE 6-20.
 - A.- A BRIGHT AND CLEAN COCKPIT.
 - B.- A CLEAN EXTERIOR SURFACE.
 - C.- ALL AIRPLANE IN GOOD CONDITION.
 - D.- A CLEAN HOUSE.
- 17.- IN THE CONVERGING PART OF THE VENTURI, WHAT HAPPEN WITH VELOCITY AND PRESSURE? (18705) REF.: FAA-H-8083-30, PAGE 3-40.
 - A.- VELOCITY AND STATIC PRESSURE WOULD INCREASE.
 - B.- THE VELOCITY AND STATIC PRESSURE WOULD DECREASE.
 - C.- VELOCITY WOULD DECREASE AND STATIC PRESSURE WOULD INCREASE.
 - D.- VELOCITY WOULD INCREASE AND STATIC PRESSURE WOULD DECREASE.
- 18.- WHAT IS BASICALLY INSTRUMENTATION? (18805) REF.: AC 65-15A, PAGE 469.
 - A.- THE SCIENCE OF MEASUREMENT.
 - B.- THE SCIENCE OF INDICATION.
 - C.- THE SCIENCE OF SHOWING DATA.
 - D.- INSTRUMENTS AND ADVERTISING.
- 19.- WHAT DO YOU HAVE TO USE TO INSPECT FOR CORROSION IN AN AIRPLANE AND BE SURE THAT NO AREA IS LEFT UNINSPECTED? (18777) REF.: FAA-H-8083-30, PAGE 6-8.
 - A.- INSTRUMENT.
 - B.- CHECKLISTS.
 - C.- TWO OR THREE METHODS.
 - D.- LEVEL TWO MECHANIC.

20.- WHERE ARE POWER PLANTS USUALLY MOUNTED IN MULTI-ENGINE HELICOPTERS? (18834) REF.: AC 65-15A, PAGE 24.

- A.- THEY ARE MOUNTED IN SEPARATE ENGINE NACELLES.
- B.- THEY ARE MOUNTED IN THE SAME NACELLE.
- C.- THEY ARE MOUNTED ONE BACK AND ONE FORWARD.
- D.- THEY ARE MOUNTED IN TANDEM.

21.- IN GENERAL, WHAT WILL BE THE FLASHPOINT FOR SOLVENT CLEANERS USED IN AIRCRAFT CLEANING? (18797) REF.: FAA-H-8083-30, PAGE 6-23.

- A.- NO MORE THAN 105°F.
- B.- NO LESS THAN 105°F.
- C.- 105°F EXACTLY.
- D.- SOLVENT CLEANERS DO NOT HAVE FLASHPOINT.

22.- THE PRIMARY GROUP OF FLIGHT CONTROL SURFACES CONSISTS OF (18830) REF.: AC 65-15A, PAGE 18.

- A.- AILERONS, ELEVATORS AND RUDDERS.
- B.- FLAPS, TRIM TABS AND STABILIZERS.
- C.- SPEED BRAKES, LONGERONS AND ELEVATORS.
- D.- WING TIPS, GROUND SPOILERS AND RIBS.

23.- WHAT HAPPENS IN A TWO-BLADED SYSTEM HELICOPTER? (19032) REF.: AC 65-15A, PAGE 53.

- A.- THE BLADES FLAP AS A UNIT.
- B.- THE BLADES FLAP INDEPENDENTLY.
- C.- THE BLADES DO NOT FLAP.
- D.- THE BLADES FLAP ONLY OCCASIONALY.

24.- WHAT HAPPENS IN A HOVERING FLIGHT, IF LIFT AND THRUST ARE LESS THAN WEIGHT AND DRAG? (18987) REF.: AC 65-15A, PAGE 50.

- A.- THE HELICOPTER ENGINE IDLES.
- B.- THE HELICOPTER STAYS IN A HOVERING FLIGHT.
- C.- THE HELICOPTER DESCENDS VERTICALLY.
- D.- THE HELICOPTER ASCENDS VERTICALLY.

25.- WHAT ARE USED IN SOME SYSTEMS IN ADDITION TO TURNBUCKLES? (19031) REF.: AC 65-15A, PAGE 65.

- A.- WIRE CONNECTORS, PULLEYS AND SHROUD MUST BE USED.
- B.- CABLE CONNECTORS AND PULLEYS ARE ALWAYS USED.
- C.- CABLE CONNECTORS ARE NEVER USED.
- D.- CABLE CONNECTORS ARE USED.

26.- WHAT MAY BE WRONG IF THE POINTER FAILS TO RESPOND ENTIRELY? (19272) REF.: AC 65-15A, PAGE 474.

- A.- THE GLASS COVER MAY BE BROKEN.
- B.- THE TRANSMISION MECHANISM IS DISCONNECTED.
- C.- THE MECHANISM IS, IN ALL PROBABILITY, DEFECTIVE.
- D.- THE ELECTRICAL POWER IS CUT OFF.

27.- WHAT MUST BE DONE IN ADDITION TO ROUTINE MAINTENANCE INSPECTIONS IN AMPHIBIANS OR SEAPLANES? (18778) REF.: FAA-H-8083-30, PAGE 6-8.

- A.- COULD BE CHECKED OCCASSIONALLY AND CRITICAL AREAS CLEANED OR TREATED EVERY DAY.
- B.- MAY BE CHECKED MONTHLY AND CRITICAL AREAS WIPED OR TREATED, AS NECESSARY.
- C.- SHOULD BE CHECKED DAILY AND CRITICAL AREAS CLEANED OR TREATED, AS NECESSARY.
- D.- SHOULD BE CHECKED WEEKLY AND CRITICAL AREAS PAINTED, CLEANED OR TREATED, AS NECESSARY.

28.- WHAT IS THE SIMPLEST FORM OF AN ATOM? (18929) REF.: FAA-H-8083-30, PAGE 10-2.

- A.- THE NITROGEN ATOM.
- B.- THE COPPER ATOM.
- C.- THE HYDROGEN ATOM.
- D.- THE OXYGEN ATOM.

- 29.- WHAT QUALITY HAVE THE HUMAN WASTE PRODUCTS AND THE CHEMICALS USED IN LAVATORIES? (18781) REF.: FAA-H-8083-30, PAGE 6-9.
 - A.- ARE VERY EXPENSIVE AND DIFFICULT TO GET.
 - B.- ARE VERY CORROSIVE TO COMMON AIRCRAFT METALS.
 - C.- SMELL VERY BAD AND ARE CHEAP TO BUY.
 - D.- ARE VERY DANGEROUS TO COMPOSITES.
- 30.- HOW OFTEN IS THE INSPECCION FOR CORROSION IN ANY AIRCRAFT RECOMENDED? (18779) REF.: FAA-H-8083-30, PAGE 6-8.
 - A.- ANNUALLY.
 - B.- IN ANY INSPECTION.
 - C.- WEEKLY.
 - D.- DAILY BASIS.
- 31.- HOW LONG HAVE HIGH TENSION IGNITION SYSTEMS BEEN IN USE? (19298) REF.: AC 65-12A, PAGE 186.
 - A.- FOR MORE THAN HALF A CENTURY.
 - B.- FOR MORE THAN A CENTURY.
 - C.- FOR MORE THAN SIXTY YEARS.
 - D.- FOR MANY TIMES.
- 32.- HOW IS THE WEIGHT OF A SOLID BODY SUBMERGED IN A LIQUID OR A GAS TO BE IT IN FREE SPACE? (18701) REF.: FAA-H-8083-30, PAGE 3-28.
 - A.- IN THE LIQUID IT WEIGHS LESS THAN IN THE FREE SPACE.
 - B.- IN THE LIQUID IT WEIGHS MORE THAN IN THE FREE SPACE.
 - C.- IN THE LIQUID IT WEIGHS EQUAL THAN IN THE FREE SPACE.
 - D.- IN THE LIQUID IT WEIGHS 10% LESS THAN IN THE FREE SPACE.
- 33.- WHAT MUST A HELICOPTER STRUCTURAL MEMBERS RESIST? (18837) REF.: AC 65-15A, PAGE 25.
 - A.- SHOCK.
 - B.- LOADS.
 - C.- FORCE.
 - D.- STRESS.

34.- WHAT IS HEAT? (18696) REF.: FAA-H-8083-30, PAGE 3-19.

- A.- IS A FORM OF ENERGY.
- B.- IS A FORM OF TEMPERATURE.
- C.- IS A FORM OF RADIATION.
- D.- IS A FORM OF WORK.

35.- WHAT HAPPENS IF HEAT IS APPLIED TO A METAL? (18729) REF.: FAA-H-8083-30, PAGE 5-2.

- A.- WILL CAUSE IT TO CONTRACT OR BECOME LARGER.
- B.- WILL CAUSE IT TO EXPAND OR BECOME LARGER.
- C.- WILL CAUSE IT TO EXPAND OR BECOME SHORTER.
- D.- WILL CAUSE IT TO INCREASE THE WEIGHT OR BECOME HEAVIER.

36.- WHAT SEWING METHOD IS NECESSARY TO CLOSE THE FINAL OPENING IN THE COVERING? (18846) REF.: AC 65-15A, PAGE 90.

- A.- MACHINE SEWING.
- B.- DOUBLE STITCH SEWING.
- C.- AUTOMATIC SEWING.
- D.- HAND SEWING.

37.- WHAT IS THE RIVETS GRIP LENGTH? (18761) REF.: FAA-H-8083-30, PAGE 5-64.

- A.- THE TOTAL SHEET THICKNESS PLUS THE RIVET LENGTH.
- B.- THE NOMINAL RIVET LENGTH.
- C.- THE SHANK DIAMETER.
- D.- THE MAXIMUN TOTAL SHEET THICKNESS TO BE RIVETED.

38.- WHAT ARE THE METHODS TO COVER FUSELAGES? (18851) REF.: AC 65-15A, PAGE 95.

- A.- THE SLEEVE OR BLANKET METHOD.
- B.- THE FABRICS SLEEVE OR PLASTIC BLANKET METHOD.
- C.- THE PARTIAL O TOTAL COVERING METHOD.
- D.- THER ARE NO METHODS TO COVER FUSELAGES.

39.- IN WHAT IS NOT BASED THE MINIMUM FUEL FOR TURBINE ENGINE POWERED AIRCRAFT? (18718) REF.: FAA-H-8083-30, PAGE 4-5.

- A.- ON ENGINE SPEED.
- B.- ON ENGINE HORSEPOWER.
- C.- ON ALTITUDE OF FLIGHT.
- D.- ON FLIGHT DISTANCE.

40.- WHAT MUST BE MAINTAINED FOR EFFICIENT OPERATION OF SHOCK STRUTS? (19271) REF.: AC 65-15A, PAGE 345.

- A.- THE PROPER CLEANNESS AND CORRECT O-RING.
- B.- THE PROPER FLUID LEVEL AND AIR PRESSURE.
- C.- THE PROPER TIRE PRESSURE AND FLUID LEVEL.
- D.- THE PROPER WHEEL, BRAKES AND FLUID.

41.- WHAT FORCES ACT ON AN AIRCRAFT, WHETHER IT IS ON THE GROUND OR IN FLIGHT? (19042) REF.: AC 65-15A, PAGE 131.

- A.- THE FORCES ARE CUTING, PUSHING OR BENDING.
- B.- THE FORCES ARE PULLING. PUSHING OR TWISTING.
- C.- THE FORCES ARE WEIGHT, THRUST OR DRAG.
- D.- THE FORCES ARE LIFTING, PUSHING OR DOWNWARD.

42.- HOW MUST THE MIXTURE BE IN ORDER FOR AN ENGINE TO DEVELOP MAXIMUN POWER AT FULL THROTTLE? (19115) REF.: AC 65-12A. PAGE 121.

- A.- THE FUEL MIXTURE MAY BE RICHER THAN FOR CRUISE.
- B.- THE FUEL MIXTURE IS THE SAME ALL THROUGH THE FLIGHT.
- C.- THE FUEL MIXTURE MUST BE RICHER THAN FOR CRUISE.
- D.- THE FUEL MIXTURE MUST BE LEANER THAN FOR CRUISE

43.- WHAT ARE THE FACTORS TO CONSIDER IN THE SELECTION OF THE CORRECT RIVET FOR INSTALLATION? (18757) REF.: FAA-H-8083-30, PAGE 5-61.

- A.- INSTALLATION LOCATION, PROTECTION OF THE MATERIAL BEING RIVETED, DIAMETERS OF THE MATERIAL BEING RIVETED AND STRESS DESIRED.
- B.- INSTALLATION PLACE, COMPOSITION OF THE TOOLS BEING USED, THINNESS OF THE MATERIAL BEING RIVETED AND STRENGTH DESIRE.
- C.- WORK LOCATION, ILUMINATION OF THE PLACE, COMPOSITION OF THE MATERIAL BEING RIVETED, THICKNESS OF THE MATERIAL BEING RIVETED AND QUALITY OF THE TOOLS.
- D.- INSTALLATION LOCATION, COMPOSITION OF THE MATERIAL BEING RIVETED, THICKNESS OF THE MATERIAL BEING RIVETED AND STRENGTH DESIRED.

44.- WHAT IS THE ADVANTAGE OF EXTERNAL AIRCRAFT SURFACES WITH REGARD TO INSPECTION AND MAINTENANCE? (18782) REF.: FAA-H-8083-30, PAGE 6-9.

- A.- IT HAS NO SPECIAL ADVANTAGES.
- B.- VERY EASY TO READ.
- C.- READILY VISIBLE AND ACCESSIBLE.
- D.- DEPENDS ON THE AIRCRAFT TYPE.

45.- WHAT IS NEEDED IN ORDER TO PRODUCE EXTENSIVE PITTING DAMAGE IN DISSIMILAR METAL PARTS? (18773) REF.: FAA-H-8083-30, PAGE 6-4.

- A.- THE PRESENCE OF SALT.
- B.- THE PRESENCE OF A CONDUCTOR.
- C.- THE ABSENCE OF ZINC CHROMATE PRIMER.
- D.- BIG CHANGE IN TEMPERATURES.

46.- WHAT MAY HAPPEN IF THE PROPELLER BLADES HAVE CRACKS? (18799) REF.: FAA-H-8083-30, PAGE 6-23.

- A.- MAY NEED TO BE REWORKED.
- B.- MAY NEED TO BE REJECTED.
- C.- MAY TEND TO OXIDIZE.
- D.- MAY BE CUT.

47.- WHAT MUST EVERY PRINT HAVE? (18649) REF.: FAA-H-8083-30, PAGE 2-4.

- A.- THE BEST INK.
- B.- SOME MEANS OF IDENTIFICATION.
- C.- SOME MEANS OF ISSUE.
- D.- A POOR PAPER.

48.- WHAT MUST ELEMENTS HAVE IN ORDER TO BE GOOD CONDUCTORS? (18931) REF.: FAA-H-8083-30, PAGE 10-3.

- A.- MANY FREE ELECTRONS.
- B.- VERY LOW WEIGHT.
- C.- BE VERY SOFT.
- D.- BE VERY EXPENSIVE.

49.- WHAT IS REQUIRED FOR THE ELECTRONS TO STAY IN AN ORBIT? (18930) REF.: FAA-H-8083-30, PAGE 10-2.

- A.- A HIGH TEMPERATURE.
- B.- A CERTAIN AMOUNT OF ELECTRICITY.
- C.- A CERTAIN AMOUNT OF HEAT.
- D.- A CERTAIN AMOUNT OF ENERGY.

50.- WHAT ARE THE FORCES THAT ACT IN THE HELICOPTER DURING FLIGHT? (18979) REF.: AC 65-15A, PAGE 49.

- A.- LIFT, THRUST, WEIGHT AND DRAG.
- B.- CUT, TORSION, FORWARD AND AFT.
- C.- BENDING, COMPRESSION, SHEAR AND TORSION.
- D.- BACK, FORWARD, UP AND DOWN.

51.- WHAT IS THE LINK BETWEEN THE ENGINEERS WHO DESIGN AN AIRCRAFT AND THE WORKERS WHO BUILD, MAINTAIN, AND REPAIR IT? (18647) REF.: FAA-H-8083-30, PAGE 2-1.

- A.- THE HANGARS, TOOLS, TECHNICAL ORDERS AND LADDERS.
- B.- THE HANGARS AND LADDERS.
- C.- THE TOOLS AND TECHNICAL ORDERS.
- D.- THE DRAWINGS AND PRINTS.

52.- WHAT MAY DAMAGE THE DRACON? (18848) REF.: AC 65-15A, PAGE 93.

- A.- THE EXCESSIVE WATER.
- B.- THE EXCESSIVE HEAT.
- C.- THE EXCESSIVE SUN.
- D.- THE EXCESSIVE WIND.

53.- WHAT CREATES THE DISSYMMETRY OF LIFT IN AN HELICOPTER? (18989) REF.: AC 65-15A, PAGE 51.

- A.- THE DIFFERENTIAL TORQUE BETWEEN TWO ROTORS.
- B.- THE WIND DURING HOVERING OR VERTICAL FLIGHT.
- C.- THE HORIZONTAL FLIGHT OR THE DIFFERENTIAL TILT.
- D.- THE HORIZONTAL FLIGHT OR WIND DURING HOVERING.

54.- WHAT IS AN IMPORTANT CONSIDERATION WHEN CHOOSING MATERIAL TO USE IN AIRPLANE PARTS? (18728) REF.: FAA-H-8083-30, PAGE 5-1.

- A.- THE DENSITY OF MATERIALS.
- B.- THE VOLUME OF MATERIALS.
- C.- THE HARDNESS.
- D.- THE STRENGTH.

55.- WHAT SIDE HAVE TO BE DEBURRED DURING A TUBING FLARING JOB? (18808) REF.: FAA-H-8083-30, PAGE 7-6.

- A.- BOTH ENDS BY THE INSIDE AND THE OUTSIDE TUBING.
- B.- THE OUTSIDE.
- C.- THE INSIDE.
- D.- THE INSIDE AND THE OUTSIDE.

56.- IN HOW MANY GROUP ARE DAMAGES CLASSIFIED? (19041) REF.: AC 65-15A, PAGE 131.

- A.- FROM SMALL DENTS TO HOLES.
- B.- FROM NEGLIGIBLE TO BIG DAMAGES.
- C.- INTO FOUR SPECIFIC GROUPS.
- D.- INTO FOUR GENERAL GROUPS.

57.- WHENEVER POSSIBLE, HOW SHOULD DAMAGED TUBING AND FLUID LINES BE REPAIRED? (18803) REF.: FAA-H-8083-30, PAGE 7-2.

- A.- SHOULD BE REPAIRED WITH GOOD PARTS.
- B.- SHOULD BE REPAIRED WITH BRIGHT PARTS.
- C.- SHOULD BE REPAIRED WITH NEW PARTS.
- D.- SHOULD BE REPAIRED WITH OVERHAULED PARTS.

58.- WHAT DOES A COMPRESSION FORCE TRY TO DO? (18670) REF.: FAA-H-8083-30, PAGE 3-14.

- A.- IT TRIES TO CRUSH AN OBJECT.
- B.- IT TRIES TO TWIST AN OBJECT.
- C.- IT TRIES TO SLICE OR TO CUT AN OBJECT.
- D.- IT TRIES TO STRESS AN OBJECT.

59.- IN WHAT BODY PARTS OF THE BODY CAN COMPOSITE PRODUCTS BE VERY HARMFUL? (18740) REF.: FAA-H-8083-30, PAGE 5-33.

- A.- IN THE SKIN. EYES AND LUNGS.
- B.- IN THE HAND, FEET AND HAIR.
- C.- IN THE CHEST, ARMS AND FACE.
- D.- IN THE ELBOW, FINGERS AND NOSE.

60.- WHAT EFFECT CAN GREASE AND DIRT ACCUMULATION PRODUCE ON THE AIR-COOLED ENGINE? (18794) REF.: FAA-H-8083-30, PAGE 6-23.

- A.- PRODUCE A COOLING EFFECT.
- B.- PRODUCE A REFRIGERATING EFFECT.
- C.- PRODUCE AN INSULATION EFFECT.
- D.- PRODUCE A WEIGHT INCREASE.

61.- WHY IS ZINC CHROMATE PRIMER APPLIED TO METALLIC SURFACES? (18847) REF.: AC 65-15A, PAGE 115.

- A.- BECAUSE THE COVERING IS CORROSION RESISTANT.
- B.- BECAUSE THE COVERING HAS A NICE COLOR.
- C.- BECAUSE IT PROTECTS IT FROM THE SUN.
- D.- BECAUSE THE COVERING IS ENAMEL OR LACQUER.

62.- WHICH ARE CORROSION RESISTANT METALS? (18731) REF.: FAA-H-8083-30, PAGE 5-4.

- A.- NICKEL STEELS OR ALUMINUM METALS.
- B.- CHROME-VANADIUM OR CHROMIUM-IRON METALS.
- C.- ALUMINUM OR COPPER METALS.
- D.- CHROME-NICKEL OR STAINLESS STEELS METALS.

63.- WHAT PRECAUTIONS MUST BE TAKEN WHEN CLEANING ASSEMBLED AIRCRAFT WITH CHEMICAL CLEANERS? (18801) REF.: FAA-H-8083-30, PAGE 6-25.

- A.- THEY MUST BE USED WITH GREAT CARE.
- B.- THEY MUST BE USED IN OPEN PLACES.
- C.- THEY MUST NOT BE USED WITH SUN LIGHT.
- D.- THEY MUST BE USED WITH MASK AND SPECIAL WORK CLOTH.

64.- WHAT IS THE COMMON NAME FOR THE CORROSION BETWEEN DIFFERENT METALS? (18789) REF.: FAA-H-8083-30, PAGE 6-16.

- A.- ELECTROLYTIC OR DISSIMILAR METALS CORROSION.
- B.- INTERGRANULAR OR DISSIMILAR METALS CORROSION.
- C.- CHEMICAL OR DISSIMILAR METALS CORROSION.
- D.- SPOT WELDING OR SIMILAR METALS CORROSION.

65.- WHAT DO YOU HAVE TO CHECK BEFORE USING THE GRINDER? (19128) REF.: AC 65-15A, PAGE 141.

- A.- THE BENCH GRINDER.
- B.- THE ABRASIVE GRAIN.
- C.- THE WHEEL FOR CRACKS.
- D.- THE ELECTRICAL POWER.

66.- WHAT TYPE OF ENERGY IS VELOCITY OF THE AIR? (18702) REF.: FAA-H-8083-30, PAGE 3-32.

- A.- POTENTIAL ENERGY.
- B.- KINETIC ENERGY.
- C.- WORK ENERGY.
- D.- STATIC ENERGY.

67.- WHAT DO YOU INSPECT IN A SCRIBER BEFORE USING IT? (18866) REF.: FAA-H-8083-30, PAGE 9-21.

- A.- THE CONDITION.
- B.- THE SERIAL NUMBER.
- C.- THE POINTS FOR SHARPNESS.
- D.- THE LENGTH.

68.- WHAT MUST THE MECHANIC DO BEFORE INSTALLING A LINE ASSEMBLY IN AN AIRCRAFT? (18810) REF.: FAA-H-8083-30, PAGE 7-13.

- A.- INSPECT THE LINE AND FITTING CAREFULLY.
- B.- INSPECT THE LINE CAREFULLY.
- C.- INSPECT THE LINE, TOOLS AND FITTING CAREFULLY.
- D.- INSPECT THE LINE, FITTING, MAINTENANCE MANUAL AND TOOLS CAREFULLY.

69.- WHAT INSPECTION SHOULD BE ACCOMPLISHED AT THE TORQUE WRENCH BEFORE EACH USE? (18860) REF.: FAA-H-8083-30, PAGE 9-7.

- A.- VISUAL INSPECTION FOR DAMAGE.
- B.- VISUAL INSPECTION FOR CALIBRATION.
- C.- VISUAL INSPECTION FOR CLEANNESS.
- D.- CHECK THE INSTRUMENT READING.

70.- WHEN A HELICOPTER IS BEING WEIGHED, WHAT LOCATION MUST BE KNOWN? (18645) REF.: FAA-H-8083-30, PAGE 4-27.

- A.- THE LONGITUDINAL WEIGHING POINT.
- B.- THE LONGITUDINAL AND LATERAL WEIGHING POINTS.
- C.- THE LATERAL WEIGHING POINT.
- D.- THE LONGITUDINAL, VERTICAL AND LATERAL WEIGHING POINTS.

71.- WHAT IS THE BEST BARRIER BETWEEN METAL AND CORROSION? (18741) REF.: FAA-H-8083-30, PAGE 6-19.

- A.- THE USE OF BEST CLEANING.
- B.- A GOOD PAINT FINISH.
- C.- A FIELD TREATMENT.
- D.- THE WET WASH.

72.- WHERE MUST THE MECHANIC AVOID USING CLAMPS IN FLEXIBLE HOSES? (18816) REF.: FAA-H-8083-30, PAGE 7-21.

- A.- CLOSE TO ELECTRIC LINES.
- B.- IN PRESURIZATION AREAS.
- C.- AT ANY LOCATIONS.
- D.- WHERE THE CLAMP CAN RESTRICT OR PREVENT HOSE FLEXURE.

73.- WHAT HAPPENS DURING AN AUTOROTATION IN A HELICOPTER? (18990) REF.: AC 65-15A, PAGE 54.

- A.- THE ENGINE DOES NOT SUPPLY POWER.
- B.- THE ENGINE IS IN MAINTENANCE.
- C.- THE HELICOPTER IS IN A HOVERING POSITION.
- D.- THE CONING IS DOWNWARD.

74.- WHAT MATERIAL IS USED TO BUILD A TWIST DRILL? (18863) REF.: FAA-H-8083-30, PAGE 9-12.

- A.- CROME VANADIUM STEEL AND HIGH-SPEED CARBON ALLOY STEEL.
- B.- CROME STEEL ALLOY AND HIGH-SPEED ALLOY STEEL.
- C.- CARBON STEEL AND HIGH-SPEED ALLOY IRON.
- D.- CARBON STEEL AND HIGH-SPEED ALLOY STEEL.

- 75.- HOW MANY FORMS OF CORROSION ATTACK ON ALUMINUM ALLOYS ARE PARTICULARLY SERIOUS? (18785) REF.: FAA-H-8083-30, PAGE 6-13.
 - A.- TWO.
 - B.- THREE.
 - C.- FOUR.
 - D.- FIVE.
- 76.- WHAT FLAME HAS TO BE USED FOR WELDING CHROME MOLYBDENUM? (19130) REF.: AC 65-15A, PAGE 259.
 - A.- A SOFT NEUTRAL FLAME.
 - B.- A STRONG NEUTRAL FLAME.
 - C.- A BIG STRONG FLAME.
 - D.- A MEDIUM SOFT NEUTRAL FLAME.
- 77.- WHAT USE MUST NEVER BE GIVEN TO A SCREWDRIVER? (18857) REF.: FAA-H-8083-30, PAGE 9-2.
 - A.- USE IT LIKE PLIERS OR PUNCHES.
 - B.- USE IT LIKE CHISELS OR PUNCHES.
 - C.- USE IT LIKE A FORK OR KNIFE.
 - D.- USE IT LIKE A STONE OR HAMMER.
- 78.- WHAT HAPPENS TO AIR AS ALTITUDE INCREASES? (19111) REF.: AC 65-12A, PAGE 120.
 - A.- THE AIR BECOMES DENSER.
 - B.- THE AIR BECOMES LESS DENSE.
 - C.- THE OXYGEN PROPORTION CHANGES.
 - D.- THE AIR BECOMES HEAVIER.

79.- WHAT IS AN AIRFOIL? (18991) REF.: AC 65-15A, PAGE 30.

- A.- AN AIRFOIL IS AN AIRPLANE DESIGNED TO OBTAIN A DESIRABLE REACTION FROM THE AIR THROUGH WHICH IT MOVES.
- B.- AN AIRFOIL IS A SURFACE DESIGNED TO OBTAIN A DESIRABLE REACTION FROM THE AIR THROUGH WHICH IT MOVES.
- C.- AN AIRFOIL IS A SURFACE DESIGNED TO OBTAIN A DESIRABLE REACTION FROM THE AIR WHEN IT IS ON THE GROUND.
- D.- AN AIRFOIL IS A PLANE CONSTRUCTED TO OBTAIN A DESIRABLE REACTION FROM THE AIR THROUGH WHICH IT MOVES.

80.- WHAT DOES A PYLON USUALLY HAVE? (18829) REF.: AC 65-15A, PAGE 24.

- A.- BULKHEADS, COVERS, DOORS, RIVETS AND BOLTS WITH NUTS.
- B.- SUPPORTS, FRAMES, FUEL PUMP AND PIPES.
- C.- ALUMINUM, ALLOYS, GASKETS AND BEAM
- D.- BULKHEADS, FORMERS, FRAMES, STRINGERS AND BEAMS.

81.- WHAT HAS TO SHOW A PROPERLY DESIGNED JOINT WELD? (18856) REF.: FAA-H-8083-30, PAGE 8-33.

- A.- UNIFORM IN WIDTH.
- B.- THE BASE METAL IS OVERHEATING.
- C.- THE EDGE OF THE BEAD IS NOT IN A STRAIGHT LINE.
- D.- THE PENETRATION SHOWS GAS POCKETS.

82.- WHEN AN AIRCRAFT IS IN A STATE OF EQUILIBRIUM? (19289) REF.: AC 65-15A. PAGE 37.

- A.- WHEN THE SUM OF ALL THE FORCES ACTING ON AN AIRCRAFT AND ALL THE WEIGHT IS EQUAL TO ZERO.
- B.- WHEN THE SUM OF ALL ITEMS ON AN AIRCRAFT AND ALL THE MOMENTS IS EQUAL TO ZERO.
- C.- WHEN THE AIRCRAFT IS IN STABILITY.
- D.- WHEN THE SUM OF ALL THE FORCES ACTING ON AN AIRCRAFT AND ALL THE MOMENTS IS EQUAL TO ZERO.

83.- WHAT IS A MACHINE? (18661) REF.: FAA-H-8083-30, PAGE 3-8.

- A.- ANY DEVICE THAT TRANSFORMS ENERGY.
- B.- ALL DEVICES THAT USE FUEL.
- C.- ANY DEVICE WITH WHICH WORK MAY BE ACCOMPLISHED.
- D.- ANY ELEMENT WITH WHICH WORK MAY BE ACCOMPLISHED.

84.- WHAT DO YOU HAVE TO USE WHILE YOU ARE DRILLING? (19126) REF.: AC 65-15A, PAGE 139.

- A.- SAFETY GOGGLES.
- B.- SAFETY GLASSES.
- C.- WORK WEAR.
- D.- CLEAN TOOLS.

85.- WHAT DOES THIS ADVISORY CIRCULAR RECOMEND FOR WORKING DURING AN ENGINE OVERHAUL? (19310) REF.: AC 65-12A. PAGE 412.

- A.- ALWAYS USE THE PROPER TOOL FOR THE JOB AND THE ONE THAT FITS.
- B.- USE THE PROPER TOOL FOR THE JOB AND KEEP IT CLEAN.
- C.- SOME TIME USE THE PROPER TOOL FOR THE JOB AND THE ONE THAT IS NEW.
- D.- DRAIN THE ENGINE OIL SUMPS AND CHANGE THE OIL FILTERS.

86.- WHAT OCCURS IF A LIGHTNING STRIKES AN AIRCRAFT? (18820) REF.: FAA-H-8083-30, PAGE 8-17.

- A.- THE ELECTRICAL CURRENT MUST BE ELIMINATED.
- B.- THE ELECTRICAL CURRENT MUST BE CONDUCTED THROUGH THE DISCHARGER.
- C.- THE ELECTRICAL CURRENT MUST BE CONDUCTED THROUGH THE ELECTRICAL SYSTEMS.
- D.- THE ELECTRICAL CURRENT MUST BE CONDUCTED THROUGH THE STRUCTURE.

87.- WHY DO YOU HAVE TO INSPECT THE SURROUNDING AREA DURING A STRUCTURAL DAMAGE? (19124) REF.: AC 65-15A, PAGE 127.

- A.- FOR LOST OF PAINT.
- B.- FOR RIVETS DAMAGE.
- C.- FOR EVIDENCE OF CORROSION.
- D.- FOR EVIDENCE OF BIGGEST DAMAGE.

88.- HOW MANY SCALES DOES A MICROMETER HAVE? (18868) REF.: FAA-H-8083-30, PAGE 9-23.

- A.- TWO.
- B.- FIVE.
- C.- FOUR.
- D.- THREE.

89.- WHAT DO ALL HEAT ENGINES HAVE IN COMMON? (19101) REF.: AC 65-12A, PAGE 1.

- A.- THE ABILITY TO CONVERT HEAT ENERGY INTO MECHANICAL ENERGY.
- B.- THE ABILITY TO CONVERT MECHANICAL ENERGY INTO HEAT ENERGY.
- C.- THE ABILITY TO USE FUEL AND TO DELIVER POWER.
- D.- THE ABILITY TO SUPPORT AIRPLANES.

90.- WHAT HAVE TO DO AFTER ASSEMBLY ALL FLEXIBLE HOSES? (18815) REF.: FAA-H-8083-30, PAGE 7-18.

- A.- DEPEND THE PRESSURE THAT WILL SUPPORTED.
- B.- COULD BE RE-CHEQUED.
- C.- MUST BE PROOF-TESTED.
- D.- MAY BE INSPECTED.

91.- WHAT CHARGE DO ELECTRONS POSSESS? (18977) REF.: FAA-H-8083-30, PAGE 10-4.

- A.- DEPEND ON THE CURRENT FLOW DIRECTION.
- B.- NEUTRAL CHARGE.
- C.- NEGATIVE CHARGE.
- D.- POSITIVE CHARGE.

92.- WHAT IS THE NAME OF THE PHENOMENON THAT CAUSES FLOW CHANGE? (19029) REF.: AC 65-15A, PAGE 59.

- A.- SUPERSONIC FLOW.
- **B.- WAVE FORMATIONS.**
- C.- WAVE COMPRESSION.
- D.- FLOW DIRECTION.

93.- WHAT DO WE HAVE TO DO WHEN WE FIND DEEP PIT IN THE TEETH OF A GEAR? (19311) REF.: AC 65-12A, PAGE 413.

- A.- SEND IT TO MANUFACTURER.
- B.- REJECT IT.
- C.- REWORK IT.
- D.- CHANGE THE TEETH.

94.- WHAT TYPE OF CONTINUOUS MAINTENANCE PROGRAM UTILIZE THE AIRLINES? (18818) REF.: FAA-H-8083-30, PAGE 8-15.

- A.- A 100 HOURS AND ANNUAL INSPECTIONS.
- B.- A ROUTINE AND DETAILED INSPECTIONS.
- C.- PROGRESSIVE ANNUAL INSPECTIONS.
- D.- A DIFFERENT LEVEL OF INSPECTIONS.

95.- WHAT IS THE PRINCIPAL FLUID USED FOR PROPULSION IN EVERY TYPE OF POWERPLANT EXCEPT THE ROCKET? (19103) REF.: AC 65-12A, PAGE 1.

- A.- NAPHTA.
- B.- PETROLEUM.
- C.- AIR.
- D.- FUEL.

96.- WHAT ARE THE GROUPS THAT NUTS CAN BE DIVIDED INTO? (18747) REF.: FAA-H-8083-30, PAGE 5-45.

- A.- ALUMINUM AND NON -ALUMINUM NUTS.
- B.- METAL AND NON-METAL NUTS.
- C.- NON-SELF-LOCKING AND SELF LOCKING NUTS.
- D.- BIG AND SMALL NUTS.

97.- WHICH IS THE BASIC COMPONENT OF A CABLE? (18766) REF.: FAA-H-8083-30, PAGE 5-75.

- A.- THE WIRE.
- B.- THE ENVELOPE.
- C.- THE METAL.
- D.- THE LINE.

98.- IN WHAT HEAT TREATMENT ALLOYS EXISTS A SERIOUS POTENTIAL FIRE HAZARD? (18739) REF.: FAA-H-8083-30, PAGE 5-27.

- A.- IN ANY ALLOYS.
- B.- IN ALUMINUM ALLOYS.
- C.- IN TITANIUM ALLOYS.
- D.- IN MAGNESIUM ALLOYS.

99.- WHAT MUST BE DONE AFTER AN NDT INSPECTION? (18854) REF.: FAA-H-8083-30, PAGE 8-27.

- A.- THE PART MUST BE STORED.
- B.- THE PART MUST BE RINSED.
- C.- THE PART MUST BE DEMAGNETIZED.
- D.- THE PART MUST BE CLEANED.

100.- WHAT IS THE WINGSPAN? (18828) REF.: FAA-H-8083-30, PAGE 1-20.

- A.- THE CHORD OF THE WING.
- B.- THE LENGHT OF THE WING FROM LEADING EDGE TO TRAILING EDGE.
- C.- THE LENGHT OF THE WING FROM WINGTIP TO WINGTIP.
- D.- THE LENGHT OF THE WING FROM WINGTIP TO WINGROOT.

101.- WHAT HAPPENS WHEN THE TEETH IN A GEAR DECREASE? (18826) REF.: FAA-H-8083-30, PAGE 1-9.

- A.- THE ROTATIONAL SPEED OF THE GEAR DECREASES.
- B.- THE ROTATIONAL SPEED OF THE GEAR INCREASES.
- C.- THE ROTATIONAL SPEED OF THE GEAR IS THE SAME.
- D.- DEPENDS ON THE SPEED IN THE MAIN GEAR.

102.- WHEN CLEANING AN ENGINE, WHAT DO YOU HAVE TO DO WITH THE ENGINE COWLING? (18795) REF.: FAA-H-8083-30, PAGE 6-23.

- A.- CLOSE, OPEN OR REMOVE IT AS MUCH AS POSSIBLE.
- B.- OPEN IT AS MUCH AS POSSIBLE.
- C.- REMOVE IT COMPLETELY.
- D.- OPEN OR REMOVE IT AS MUCH AS POSSIBLE.

103.- WHEN ARE ALL THE FORCES ACTING ON THE AIRPLANE IN EQUILIBRIUM? (18707) REF.: FAA-H-8083-30, PAGE 3-43.

- A.- WHEN THE AIRPLANE IS IN CRUISE FLIGHT AT A CONSTANT VELOCITY.
- B.- WHEN THE AIRPLANE IS IN STRAIGHT-AND-LEVEL FLIGHT AT A CONSTANT VELOCITY.
- C.- WHEN THE AIRPLANE IS AT CONSTANT POWER AND CONSTANT VELOCITY.
- D.- ALWAYS.

104.- WHAT IS MOTION? (18813) REF.: AC 65-15A, PAGE 29.

- A.- IS THE ACT OR PROCESS OF CHANGING PLACE OR POSITION.
- B.- IS THE MOTION AROUND ANOTHER OBJECT.
- C.- IS WHEN THE AIR FLOW PASSES THROUGH AN OBJECT.
- D.- IS THE MOVEMENT OF THE AIR AROUND AND OBJECT OR THE OBJECT MOVING THROUGH THE AIR.

105.- WHAT IS ESSENTIAL TO A GOOD HEAT TREATMENT? (18736) REF.: FAA-H-8083-30, PAGE 5-16.

- A.- ACCURATE TEMPERATURE MEASUREMENT.
- B.- ACCURATE HEAT TREATMENT.
- C.- ACCURATE TEMPERATURE CONTROL.
- D.- A BIG HEAT-TREATING FURNACE.

106.- WHAT IS THE MOST DIFFICULT METAL TO PROTECT? (18788) REF.: FAA-H-8083-30, PAGE 6-15.

- A.- MAGNESIUM.
- B.- ALUMINUM.
- C.- IRON.
- D.- COPPER.

107.- WHAT EXPERIENCES AN AIRPLANE IN FLIGHT WHEN THE AERODYNAMIC LIFT FORCE ON THE WING TRIES TO RAISE THE WING? (18673) REF.: FAA-H-8083-30, PAGE 3-14.

- A.- A LIFT FORCE.
- B.- A CUT FORCE.
- C.- A BENDING FORCE.
- D.- A TORSION FORCE.

108.- WHAT IS WEIGHT? (18652) REF.: FAA-H-8083-30, PAGE 3-1.

- A.- IS A MEASURE OF THE PULL OF GRAVITY ACTING ON THE MASS OF AN OBJECT.
- B.- IS A MEASURE OF THE FORCE ACTING ON THE MASS OF AN OBJECT.
- C.- IS A MEASURE OF THE DISTANCE BETWEEN TWO OR MORE OBJECTS.
- D.- IS THE EARTH EFFECT OVER ALL OBJECTS THAT HAVE A MASS.

109.- WHAT HAPPENS ABOUT THE AXES WHEN AN AIRCRAFT CHANGES ITS ATTITUDE IN FLIGHT? (19288) REF.: AC 65-15A, PAGE 35.

- A.- THE AIRPLANE MUST TURN ABOUT ONE AXIS ONLY.
- B.- THE AXES CHANGE POSITION.
- C.- THE AIRPLANE MUST TURN ABOUT ONE OR MORE OF THREE AXES.
- D.- THE CENTER OF GRAVITY (C.G.) CHANGES POSITION.

110.- HOW IS THE WEIGHT AND BALANCE FOR LARGE AIRPLANES COMPARED WITH SMALL AIRPLANES? (18726) REF.: FAA-H-8083-30, PAGE 4-30.

- A.- IS VERY DIFFERENT.
- B.- IS VERY SIMILAR.
- C.- IS ALMOST IDENTICAL.
- D.- IS IDENTICAL.

111.- WHAT TRIES TO DO THE FORCE WHEN A SHEAR IS APPLIED TO AN OBJECT? (18674) REF.: FAA-H-8083-30, PAGE 3-15.

- A.- THE FORCE TRIES TO PULL AN OBJECT APART.
- B.- THE FORCE TRIES TO TWIST AN OBJECT.
- C.- THE FORCE TRIES TO COMPRESS OR CRUSH AN OBJECT.
- D.- THE FORCE TRIES TO CUT OR SLICE THROUGH.

112.- WHAT ARE THE GRADES OF ALUMINUM WOOL USED TO CLEAN ALUMINUM SURFACES? (18986) REF.: FAA-H-8083-30, PAGE 6-25.

- A.- IMPREGNATED, POWDERED AND ACID.
- B.- COARSE, MEDIUM AND FINE.
- C.- NEUTRAL, SOFT AND HARD.
- D.- TYPE I, TYPE II AND TYPE III.

113.- WHAT IS A VERY IMPORTANT AND EXACTING PHASE OF AIRCRAFT MAINTENANCE? (18719) REF.: FAA-H-8083-30, PAGE 4-15.

- A.- FUELING.
- B.- REGULATION REQUIREMENT.
- C.- AIRCRAFT WEIGHING.
- D.- DAILY INSPECTION.

114.- WHAT MAY VERY SEVERE INTERGRANULAR CORROSION CAUSE? (18774) REF.: FAA-H-8083-30, PAGE 6-5.

- A.- MAY SOMETIMES CAUSE THAT THE PAINT DOES NOT HOLD ON THE METAL SURFACE.
- B.- MAY SOMETIMES CAUSE CHANGES IN THE SURFACE OF A METAL.
- C.- MAY SOMETIMES CAUSE THE SURFACE OF A METAL TO EXFOLIATE.
- D.- MAY SOMETIMES CAUSE THE SURFACE OF A METAL TO STRESS.

115.- WHAT HAPPENS WHEN THE FORCE OF LIFT ON AN AIRCRAFT'S WING EQUALS THE FORCE OF GRAVITY? (19038) REF.: AC 65-15A, PAGE 30.

- A.- THE AIRCRAFT MAINTAINS LEVEL FLIGHT.
- B.- THE AIRCRAFT CAN TAKEOFF.
- C.- THE AIRCRAFT LOSES LEVEL FLIGHT.
- D.- THE AIRCRAFT INCREASES THE ALTITUDE.

116.- WHEN DOES THE SERVO STOP MOVING? (19295) REF.: AC 65-15A, PAGE 512.

- A.- WHEN ALL SIGNALS ARE EQUAL IN MAGNITUDE.
- B.- WHEN THE TWO SIGNALS ARE EQUAL IN MAGNITUDE.
- C.- WHEN THE TWO SIGNALS ARE EQUAL IN INTENSITY.
- D.- WHEN THE SIGNALS ARE IN ZERO.

117.- HOW MUST A CHISEL BE HELD WHEN USING IT? (18862) REF.: FAA-H-8083-30, PAGE 9-9.

- A.- DEPEND HOW TALL IS THE MECHANIC.
- B.- HOLD IT FIRMLY WITH GLOVES AND PROTECTIVE GLASSES.
- C.- HOLD IT FIRMLY IN ONE HAND.
- D.- HOLD IT FIRMLY WITH BOTH HANDS.

- 118.- WHEN USING RIVETS OR EVEN BOLTS, CARE MUST BE TAKEN TO ENSURE THE HOLE IS NOT (18764) REF.: FAA-H-8083-30, PAGE 5-74.
 - A.- REDUCED.
 - B.- INCORRECTLY DRILLED.
 - C.- ELONGATED OR SLANTED.
 - D.- EXACTLY.
- 119.- WHAT IS ALSO A SIGNIFICANT FACTOR IN DETERMINING IF THE AIRCRAFT IS SAFE TO OPERATE? (18712) REF.: FAA-H-8083, PAGE 4-1.
 - A.- THE AIRCRAFT BALANCE.
 - B.- THE FUEL LOADED IN THE AIRCRAFT.
 - C.- THE OIL USED IN THE ENGINES.
 - D.- THE WEATHER CONDITIONS.
- 120.- WHAT WAX SHOULD BE USED FOR LACING CORD PROTECTION? (18849) REF.: AC 65-15A, PAGE 93.
 - A.- ARTIFICIAL WAX.
 - B.- NATURAL WAX.
 - C.- BEESWAX.
 - D.- LACING WAX.
- 121.- WHAT IS BECOMING LESS OF A PROBLEM WITH THE INTRODUCTION OF SEALED LEAD-ACID BATTERIES AND THE USE OF NICKEL-CADMIUM BATTERIES? (18762) REF.: FAA-H-8083-30, PAGE 6-3.
 - A.- THE SPILLED BATTERY ACID IS A LESS OF A PROBLEM.
 - B.- THE BATTERY DRAINAGE IS LESS OF A PROBLEM.
 - C.- THE BATTERY MAINTENANCE IS A LESS OF A PROBLEM.
 - D.- THE BATTERY COST IS A LESS OF A PROBLEM.

122.- WHAT MUST BE KNOWN WHEN AN AIRCRAFT IS BEING WEIGHTED? (18638) REF.: FAA-H-8083-30, PAGE 4-16.

- A.- THE TARE AND EMPTY WEIGHT MUST BE KNOWN.
- B.- THE LEVELING POINTS MUST BE KNOWN.
- C.- THE ARMS MUST BE KNOWN.
- D.- THE SCALE MUST BE KNOWN.

123.- WHAT SHOULD BE USED WHEN A FLUID LINE PASSES THROUGH A UNIVERSAL BULKHEAD FITTING, AND IT IS DESIRED TO SECURE THE LINE TO THE BULKHEAD? (18809) REF.: FAA-H-8083-30, PAGE 7-10.

- A.- A BULKHEAD FITTING SHOULD BE USED.
- B.- A BULKHEAD FITTING MAY BE USED.
- C.- TWO OR THREE SOFT OR MEDIUM HARD CLAMPS.
- D.- RUBBER-CUSHIONED CLAMPS SHOULD BE USED.

124.- WHAT ELEMENTS CONNECT UP THE COCKPIT CONTROLS TO CONTROL CABLES AND SURFACE CONTROLS? (19290) REF.: AC 65-15A. PAGE 68.

- A.- VARIOUS MECHANICAL LINKAGES.
- B.- MANY MECHANICAL LINKAGES CABLE.
- C.- SOME MECHANICAL AND ELECTRICAL LINKAGES.
- D.- VARIOUS PUSH-PULL RODS.

125.- IN WHAT TYPE OF TUBING MAY THE TOOLS FOR ROLLING-TYPE FLARING BE USED? (18807) REF.: FAA-H-8083-30, PAGE 7-6.

- A.- IN HARD COPPER, ANY ALUMINUM, AND STEEL ALLOYS TUBING.
- B.- IN SOFT COPPER, ALCLAD OR ALUMINUM, AND BURRS TUBING.
- C.- IN SOFT COPPER, ALUMINUM, AND BRASS TUBING.
- D.- IN CORROSION RESISTANT STEEL, TITANIUM, AND BRASS TUBING.

126.- IN WHAT TYPE OF MATERIAL DOES THE ULTRASONIC INSPECTION EQUIPMENT DETECT DEFECTS? (18841) REF.: FAA-H-8083-30, PAGE 8-21.

- A.- IN ALL TYPES OF CERAMIC MATERIALS.
- B.- IN ALL TYPES OF METAL MATERIALS.
- C.- IN ALL TYPES OF MATERIALS.
- D.- IN ALL TYPES OF COMPOSITES MATERIALS.

127.- WHAT ARE THE TRIM TABS SURFACES AND WHERE ARE THEY ATTACHED? (18709) REF.: FAA-H-8083-30, PAGE 3-47.

- A.- THEY ARE SMALL FIXED SURFACES AND ARE INSTALLED TO THE WING TRAILING EDGE.
- B.- THEY ARE SOME MOVABLE CONTROLS AND ARE ATTACHED TO THE FLAPS.
- C.- THEY ARE SMALL MOVABLE SURFACES AND ARE ATTACHED TO THE TRAILING EDGE OF FLIGHT CONTROLS.
- D.- THEY ARE BIG MOVABLE SURFACES AND ARE ATTACHED TO THE LEADING EDGE OF FLIGHT CONTROLS.

128.- WHY YOU MUST NEVER APPLY COMPOUND TO THE FACE OF THE FITTING OR THE FLARE? (18836) REF.: FAA-H-8083-30, PAGE 7-13.

- A.- BECAUSE THE COMPOUND IS VERY CORROSIVE AND DANGEROUS.
- B.- BECAUSE ANY FLUID WILL DISSOLVE THE COMPOUND.
- C.- BECAUSE THE COMPOUND WILL DESTROY THE METAL-TO-METAL CONTACT.
- D.- BECAUSE THE COMPOUND WILL FORM A GUM WITH THE LIQUID.

129.- WHAT OTHER STRESSES INCLUDE THE TORSION STRESS? (18672) REF.: FAA-H-8083-30, PAGE 3-14.

- A.- BENDING AND CUT.
- B.- TENSION AND COMPRESSION.
- C.- TENSION AND TORSION.
- D.- SHEAR AND CUT.

- 130.- TORQUE IS A VERY INTERESTING CONCEPT AND OCCURRENCE, AND IT IS DEFINITELY SOMETHING THAT NEEDS TO BE DISCUSSED IN CONJUNCTION WITH? (18660) REF.: FAA-H-8083-30, PAGE 3-7.
 - A.- TOOLS AND MECHANICS.
 - B.- WORK AND POWER.
 - C.- METALS AND COMPOSITES.
 - D.- STRESS AND STRENGTH.
- 131.- WHY THE LANDING GEAR OF MOST HIGH-SPEED OR LARGE AIRCRAFT IS RETRACTED? (19281) REF.: AC 65-15A, PAGE 14.
 - A.- TO INCREASE LOAD CAPABILITY AND DECREASE VOID PLACE.
 - B.- TO INCREASE WIND RESISTANCE DURING LANDING.
 - C.- TO REDUCE NOISE AND TURBULENCE DURING ALL FLIGHT.
 - D.- TO REDUCE WIND RESISTANCE DURING FLIGHT.
- 132.- HOW MANY SYSTEMS HAS EACH CARBURATOR TO PROVIDE FOR ENGINE OPERATION UNDER VARIOUS LOADS AND AT DIFFERENT ENGINE SPEEDS? (19105) REF.: AC 65-12A, PAGE 115.
 - A.- EACH CARBURATOR HAS SIX SYSTEMS.
 - B.- EACH CARBURATOR HAS MANY SYSTEMS.
 - C.- EACH CARBURATOR HAS NINE SYSTEMS.
 - D.- THE CARBURATOR DOES NOT HAVE OTHER SYSTEMS.
- 133.- UNTIL WHEN THE MECHANICS HAVE TO TIGHTEN THE NUTS BY HAND? (18811) REF.: FAA-H-8083-30, PAGE 7-14.
 - A.- UNTIL THE NUTS ARE READY TO GIVE TORQUE.
 - B.- THE NUTS MUST BE INSTALLED WITH A SPECIAL WRENCH.
 - C.- UNTIL AN INCREASE IN RESISTANCE TO TURNING IS ENCOUNTERED.
 - D.- UNTIL THE NUT IS INSTALLED ON THE BOLT.

134.- WHAT PROVIDES THE NECESSARY TO HOLD THE AIRPLANE IN LEVEL UNACCELERATED FLIGHT? (19104) REF.: AC 65-12A, PAGE

- A.- THE FLIGHT CONTROLS.
- B.- THE DRAG.
- C.- THE LIFT.
- D.- THE ENGINE OR ENGINES.

135.- WHAT IS THE IMPORTANCE OF THE GROUND EFFECT FOR A HELICOPTER? (19024) REF.: AC 65.15A, PAGE 54.

- A.- AIDS IN INCREASING THE ENGINE POWER.
- B.- AID IN SUPPORTING THE HELICOPTER WHILE HOVERING.
- C.- THE AID IS THEORETICAL ONLY.
- D.- THE GROUND EFFECT PRODUCES ADVERSE EFFECTS.

136.- WHAT IS THE MEANING OF THE WORD HELICOPTER THAT COMES FROM GREEK? (18982) REF.: AC 65-15A, PAGE 49.

- A.- HORIZONTAL ROTATING WING AND VERTICAL ROTATING WING.
- B.- OVER CABIN WING AND TAIL WING.
- C.- HELICAL WING OR ROTATING WING.
- D.- OVER WING OR LONG ROTATING WING.

137.- WHEN OR WHERE SHOULD BE PERFORMED THE AIRCRAFT WASHING? (18792) REF.: FAA-H-8083-30, PAGE 6-20.

- A.- IN THE SHADE WHENEVER POSSIBLE.
- B.- ONLY INSIDE THE HANGAR.
- C.- DURING NIGHT.
- D.- ONLY DURING CLOUDY DAYS.

138.- WHAT COMBINATION IS THE UNIVERSAL HEAD RIVET? (18752) REF.: FAA-H-8083-30, PAGE 5-59.

- A.- THE UNIVERSAL HEAD RIVET INCLUDES ALL RIVETS TYPE.
- B.- IS A COMBINATION OF ROUNDHEAD, PLANEHEAD AND CHERRY HEAD RIVET.
- C.- IS A COMBINATION OF SQUAREDHEAD, BRIGHTHEAD AND BRAZIER HEAD RIVET.
- D.- IS A COMBINATION OF ROUNDHEAD, FLATHEAD AND BRAZIER HEAD RIVET.

139.- WHAT DETERMINES THE OVERALL LENGTH OF THE SHANK OF THE RIVET? (18755) REF.: FAA-H-8083-30, PAGE 5-61.

- A.- THE WIDTH AND HEIGH OF THE SHOP HEAD.
- B.- THE THICKNESS OF THE MATERIAL BEING RIVETED.
- C.- THE STRESS REQUIRED.
- D.- THE STRENGTH OF THE MATERIAL BEING RIVETED.

140.- WHAT CHANGES WHEN THE AIRCRAFT ATTITUDE CHANGES? (18814) REF.: AC 65-15A, PAGE 31.

- A.- THE YAW ANGLE.
- B.- THE ANGLE OF ATTACK.
- C.- THE LANDING ANGLE.
- D.- THE TRACK ANGLE.

141.- WHICH IS THE TERM APPLIED TO ALL FORMS OF GRINDING MACHINES? (19127) REF.: AC 65-15A, PAGE 140.

- A.- GRINDING WHEELS.
- B.- GRINDER.
- C.- CUTTING TOOL.
- D.- DRILL REMOVER.

142.- WHAT IS THE PRIMARY USE OF THE ENGINE TAILPIPE? (19305) REF.: AC 65-12A, PAGE 59.

- A.- TO LINE UP THE EXHAUST GASES AND TO AVOID THE TURBULENCE.
- B.- TO SAVE FUEL.
- C.- TO INCREASE THE ENGINE POWER.
- D.- TO PIPE THE EXHAUST GASES OUT OF THE AIRFRAME.

143.- WHAT AIRCRAFT PART SERVES TO CLOSE AND STREAMLINE THE AFT END OF MOST FUSELAGES? (19282) REF.: AC 65-15A, PAGE 16.

- A.- THE TAIL CONE.
- B.- THE EMPENNAGE.
- C.- THE FLIGHT CONTROLS.
- D.- THE CARGO COMPARTMENT.

144.- WHAT ALLOYS ARE USED IN THE CONSTRUCTION OF THE TAIL CONE? (18835) REF.: AC 65-15A, PAGE 24.

- A.- CHROMIUM VANADIUM ALLOY AND ALUMINUM ALLOY.
- B.- MAGNESIUM ALLOY AND ALUMINUM ALLOY.
- C.- COPPER ALLOY AND ALUMINUM ALLOY.
- D.- TITANIUM ALLOY AND MANGANESE ALLOY.

145.- WHAT ELEMENT PRODUCES LONGITUDINAL MAGNETIZATION? (18855) REF.: FAA-H-8083-30, PAGE 8-29.

- A.- A SWITCH.
- B.- A RELAY.
- C.- A SHUNT.
- D.- A SOLENOID.

146.- WHAT IS THE STUDY OF MACHINES, BOTH SIMPLE AND COMPLEX? (18659) REF.: FAA-H-8083-30, PAGE 3-4.

- A.- IS THE STUDY OF THE ENERGY OF MECHANICAL WORK.
- B.- IS THE STUDY OF THE WORK OF MECHANICAL ENERGY.
- C.- IS THE FORMULA OF THE ENERGY VERSUS MECHANICAL WORK.
- D.- IS THE TRANSFERENCE OF HEAT INTO WORK AND MECHANICAL FORCE.

147.- WHAT MAINTENANCE MUST EXECUTE IN THE CARBURETOR STRAINER? (19300) REF.: AC 65-12A, PAGE 125.

- A.- IT MUST BE REMOVED, CHECKED, CLEANED AND INSTALLED AT SCHEDULED INTERVALS.
- B.- IT MUST HAVE REGULAR CONTROLS AND CHECKS.
- C.- IT MUST BE REMOVED AND CLEANED AT SCHEDULED INTERVALS.
- D.- IT MUST BE CHANGED AT SCHEDULED INTERVALS.

148.- WHAT IS VERY IMPORTANT IN THE STUDY OF HIGH-SPEED AIRFLOW? (19027) REF.: AC 65-15A, PAGE 56.

- A.- THE WIND VELOCITY.
- B.- THE AIRFOIL.
- C.- THE AIRFLOW.
- D.- THE SPEED OF SOUND.

149.- WHAT RIVET REQUIRES SPECIAL INSTALLATION TOOLS, SPECIAL INTALLATION PROCEDURES AND SPECIAL REMOVAL PROCEDURES? (18756) REF.: FAA-H-8083-30, PAGE 5-60.

- A.- THE BULBED CHERRYLOCK RIVETS.
- B.- THE PULL-THRU RIVETS.
- C.- THE SPECIAL (BLIND) RIVETS.
- D.- THE SELF-PLUGGING RIVETS.

150.- WHAT DOES THE SKIN COVER IN AN AIRPLANE? (18819) REF.: AC 65-15A, PAGE 24.

- A.- IT COVERS THE FUSELAGE, WINGS, EMPENNAGE, NACELLES AND PODS.
- B.- IT COVERS THE FUSELAGE, WINGS, FLIGHT CONTROLS, NACELLES AND CARGO COMPARMENTS.
- C.- IT COVERS THE PILOTS, PASSENGER, CREW MEMBER AND MECHANIC.
- D.- IT COVERS THE ENTIRE AIRPLANE.

151.- WHAT IS THE SIMPLEST MACHINE, AND PERHAPS THE MOST FAMILIAR ONE? (18665) REF.: FAA-H-8083-30, PAGE 3-9.

- A.- THE LEVER.
- B.- THE GEAR.
- C.- THE WHEEL.
- D.- THE AXLE.

152.- WHAT ALLOYS ARE PRIMARILLY USED IN THE CONSTRUCTION OF THE SEMIMONOCOQUE FUSELAGE? (19276) REF.: AC 65-15A, PAGE 3.

- A.- THE ALLOYS OF ALUMINUM AND METALS.
- B.- THE ALLOYS OF ALUMINUM AND TITANIUM.
- C.- THE ALLOYS OF ALUMINUM AND MAGNESIUM.
- D.- THE ALLOYS OF ALUMINUM AND COPPER.

153.- WHAT MUST BE AVOIDED WHEN USING MECHANICAL CLEANING MATERIAL? (18800) REF.: FAA-H-8083-30, PAGE 6-25.

- A.- USING THE CLEANING MATERIAL IN CLOSE PLACE.
- B.- BREATHING OF THE CLEANING FUMES.
- C.- EYE AND SKIN CONTAMINATION.
- D.- DAMAGE TO THE FINISHES AND SURFACES.

154.- WHAT PARTS HAVE THE SELF-PLUGGING (FRICTION LOCK) BLIND RIVETS? (18754) REF.: FAA-H-8083-30, PAGE 5-60.

- A.- A RIVET HEAD WITH A HOLLOW SHANK OR SLEEVE AND A STEM THAT EXTENDS THROUGH THE HOLLOW SHANK.
- B.- A RIVET HEAD WITH A SOLID SHANK OR SLEEVE AND A STEM THAT EXTENDS THROUGH THE ENTIRE SHANK.
- C.- A PLAINHEAD RIVET WITH A HOLLOW SHANK OR SLEEVE AND A STEM THAT EXTENDS THROUGH THE SPECIAL SHANK.
- D.- A FLATHEAD RIVET WITH A COPPER SHANK OR SLEEVE AND A STEM THAT EXTENDS THROUGH THE HOLLOW SHANK.

155.- WHAT HAS TO BE CONSIDERED DURING THE SELECTION OF THE TYPE OF MATERIALS TO BE USED IN AIRCRAFT CLEANING? (18784) REF.: FAA-H-8083-30, PAGE 6-10.

- A.- THE NATURE OF THE MATTER TO BE REMOVED.
- B.- THE NATURE OF THE MATERIAL TO BE CLEANED.
- C.- THE HELP NEEDED FOR A GOOD CLEANING.
- D.- THE KNOWLEDGE OF CLEANING PERSONNEL.

156.- WHAT DOES THE RESULTANT FORCE IN A POSITIVE ANGLE OF ATTACK HAVE? (19039) REF.: AC 65-15A, PAGE 31.

- A.- IT HAS MAGNITUDE, LONGITUDE AND FORCE.
- B.- IT HAS MAGNITUDE, DIRECTION AND LOCATION.
- C.- IT HAS LIFT, DRAG AND EFFECT.
- D.- IT HAS AIRFOIL, CHORD AND LIFT.

157.- WHAT IS ONE RULE FOR THE LOCATION OF THE DATUM? (18715) REF.: FAA-H-8083-30, PAGE 4-2.

- A.- IT HAS HAVE AN EXACT MEASURE FROM THE AIRCRAFT NOSE.
- B.- IT HAS TO BE KNOWN.
- C.- IT DOES NOT CHANGE DURING THE LIFE OF THE AIRCRAFT.
- D.- IT HAS TO BE CLOSE TO THE AIRCRAFT NOSE.

158.- WHAT ARE THE MAIN TYPES OF PINS USED IN AIRCRAFT STRUCTURE? (18767) REF.: FAA-H-8083-30, PAGE 5-79.

- A.- THEY ARE THE ROLL PIN, WIRE PIN AND COTTER PIN.
- B.- THEY ARE THE TAPER PIN, FLATHEAD PIN AND COTTER PIN.
- C.- THEY ARE THE LANDING GEAR PIN, NUT PIN AND SECURITY PIN.
- D.- THEY ARE THE SAFETY CLIP PIN, ROUNDHEAD PIN AND ROLL PIN.

159.- HOW MANY WAYS OF GROUPING AIRCRAFT INSTRUMENTS ARE THERE? (18806) REF.: AC 65-15A, PAGE 469.

- A.- THERE ARE MANY WAYS OF GROUPING AIRCRAFT INSTRUMENTS.
- B.- THERE ARE TWO WAYS OF GROUPING AIRCRAFT INSTRUMENTS.
- C.- THERE ARE VARIOUS WAYS OF GROUPING AIRCRAFT INSTRUMENTS.
- D.- THE AIRCRAFT INSTRUMENTS DISTRIBUTION DEPEND THE COCKPIT SIZE.

160.- HOW MANY SIMPLE MACHINES ARE THERE? (18664) REF.: FAA-H-8083-30, PAGE 3-8.

- A.- THERE ARE ONLY TWO SIMPLE MACHINES.
- B.- THERE ARE ONLY TWENTY TWO SIMPLE MACHINES.
- C.- THERE ARE EIGHT SIMPLE MACHINES.
- D.- THERE ARE ONLY SIX SIMPLE MACHINES.

161.- WHAT FORM DO THE HEATING ELEMENTS HAVE IN THE ELECTRICITY HEATED FURNACE? (18998) REF.: FAA-H-8083-30, PAGE 5-15.

- A.- WIRE OR RIBBON.
- B.- CABLE OR TAPE.
- C.- RULE OR LEAD.
- D.- ROUND OR SQUARE.

162.- WHAT MAY RESULT, AS A GENERAL RULE, IF FURNACES ARE USED AT DIFFERENT TEMPERATURE RANGE? (18735) REF.: FAA-H-8083-30, PAGE 5-15.

- A.- THE OBJECT WILL HAVE EXTRA STRENGTH.
- B.- THE RESULTS IN WORK WILL BE OF DIFFERENT ALLOYS.
- C.- THE RESULTS IN WORK WILL BE OF SUPERIOR QUALITY.
- D.- THE RESULTS IN WORK WILL BE OF INFERIOR QUALITY.

163.- HOW MANY TYPES OF CLEANING AGENTS APPROVED FOR USE IN CLEANING AIRCRAFT ARE THERE? (18791) REF.: FAA-H-8083-30, PAGE 6-19.

- A.- VERY FEW TYPES.
- B.- HUNDREDS OF DIFFERENT TYPES.
- C.- ABOUT FOUR OR FIVE DIFFERENT TYPES.
- D.- MANY DIFFERENT TYPES.

164.- WHAT ADVANTAGES DOES THE USE OF THE SEMIMONOCOQUE FUSELAGE CONSTRUCTION HAVE? (19277) REF.: AC 65-15A, PAGE 5.

- A.- ALL CONSTRUCTION HAVE THE SAME DISADVANTAGES.
- B.- ALL CONSTRUCTION HAVE THE SAME ADVANTAGES.
- C.- IT HAS NO ADVANTAGES.
- D.- IT HAS A NUMBER OF ADVANTAGES.

165.- WHAT DO PROPELLERS OF AIRCRAFT POWERED BY RECIPROCATING OR TURBOPROP ENGINES DO WITH RESPECT TO THE AIR? (19102) REF.: AC 65-12A, PAGE 1.

- A.- ACCELERATE A SMALL MASS OF AIR THROUGH A LARGE VELOCITY CHANGE.
- B.- ACCELERATE A LARGE MASS OF AIR THROUGH A SMALL VELOCITY CHANGE.
- C.- ACCELERATE A LARGE MASS OF AIR THROUGH A LARGE VELOCITY CHANGE.
- D.- ACCELERATE A SMALL MASS OF AIR THROUGH A SMALL VELOCITY CHANGE.

166.- WHAT FURNISHES THE POWER NEEDED TO ROTATE THE PROPELLER BLADES? (19308) REF.: AC 65.12A, PAGE 325.

- A.- THE FIRE SYSTEM.
- B.- THE PISTONS.
- C.- THE ENGINE.
- D.- THE CYLINDERS.

167.- WHERE IS TYPICAL TO FIND THE PLANETARY SUN GEAR SYSTEM? (18667) REF.: FAA-H-8083-30, PAGE 3-12.

- A.- IN MANY TURBINE AND RECIPROCATING ENGINES.
- B.- IN AN ACCESORY REDUCTION GEARBOX.
- C.- IN A PROPELLER REDUCTION GEARBOX.
- D.- IN A TURBINE REDUCTION GEARBOX.

168.- WHAT MUST ALSO KNOW THE PEOPLE WHO MAINTAIN AND REPAIR AIRCRAFT? (18651) REF.: FAA-H-8083-30, PAGE 3-1.

- A.- SHOULD HAVE A KNOWLEDGE OF ONWARD PHYSICS.
- B.- SHOULD HAVE A KNOWLEDGE OF BASIC CHEMISTRY.
- C.- SHOULD HAVE A KNOWLEDGE OF BASIC METEOROLGY.
- D.- SHOULD HAVE A KNOWLEDGE OF BASIC PHYSICS.

169.- WHAT LIMITS OIL PRESSURE TO THE VALUE SPECIFIED BY THE ENGINE MANUFACTURER? (19297) REF.: AC 65-12A, PAGE 300.

- A.- THE OIL COOLER SYSTEM.
- B.- ALL THE OIL CONTROL SYSTEM.
- C.- THE OIL PRESSURE REGULATOR.
- D.- THE OIL PRESSURE RELIEF VALVE.

170.- WHICH IS THE MOST OFTEN USED LIFT-MODIFYING DEVICE, FOR SMALL AND LARGE AIRPLANE? (18710) REF.: FAA-H-8083-30, PAGE 3-48.

- A.- THE WING FLAPS AND THE LEADING EDGES.
- B.- THE MAIN FLIGHT CONTROLS.
- C.- THE ENTIRE WING.
- D.- THE WING FLAPS.

171.- WHAT DOES THE MONOSPAR WING INCORPORATE IN ITS CONSTRUCTION? (19278) REF.: AC 65-15A, PAGE 7.

- A.- ONLY ONE MAIN LONGITUDINAL MEMBER.
- B.- TWO MAIN LONGITUDINAL MEMBERS.
- C.- SEVERAL MAIN LONGITUDINAL MEMBERS.
- D.- ONE MAIN LONGITUDINAL MEMBER AND ONE SECONDARY MEMBER.

172.- WHAT DETERMINES THE RATIO OF FUEL TO AIR IN THE MIXTURE? (19106) REF.: AC 65-12A, PAGE 115.

- A.- THE AIRCRAFT SPEED.
- B.- THE MIXTURE CONTROL SYSTEM.
- C.- THE ENGINE POWER.
- D.- THE AIRCRAFT ALTITUDE AND ACTITUDE.

173.- IN WHAT APPLICATIONS CAN BE USED THE MECHANICAL LOCK TYPE OF SELF-PLUGGING RIVET? (18758) REF.: FAA-H-8083-30, PAGE 5-63.

- A.- THEY CAN BE USED IN THE SAME APPLICATIONS AS THE FRICTION LOCK RIVET.
- B.- THEY CAN BE USED IN THE SAME APPLICATIONS AS THE PULL-THRU RIVET.
- C.- THEY CAN BE USED IN THE SAME APPLICATIONS AS THE BULBED CHERRYLOCK RIVET.
- D.- THEY CAN BE USED IN THE SAME APPLICATIONS AS THE ROUNDHEAD RIVET.

174.- WHAT IS THE MAXIMUN WEIGHT OF AN AIRCRAFT? (18716) REF.: FAA-H-8083-30, PAGE 4-3.

- A.- IS THE WEIGHT OF AN AIRCRAFT SHOWED IN THE SCALE.
- B.- IS THE MAXIMUN AUTHORIZED LOAD WEIGHT OF THE AIRCRAFT AND THE FUEL.
- C.- IS THE MAXIMUN WEIGHT OF THE AIRCRAFT WEIGHED IN ANY MOMENT.
- D.- IS THE MAXIMUN AUTHORIZED WEIGHT OF THE AIRCRAFT AND ITS CONTENTS.

- 175.- WHICH AIRPLANE PARTS ARE EQUIPPED WITH AN AXLE ATTACHED TO THE LOWER CYLINDER TO PROVIDE FOR INSTALLATION OF THE WHEELS? (19270) REF.: AC 65-15A, PAGE 343.
 - A.- THE SHOCK STRUTS.
 - B.- THE FLAPS.
 - C.- THE ENGINE.
 - D.- THE MAIN FLIGHT CONTROLS.
- 176.- WHICH ARE THE MAIN STRUCTURAL PARTS OF A WING? (19279) REF.: AC 65-15A, PAGE 8.
 - A.- THE LEADING EDGE, THE TRAILING EDGE AND THE CHORD LINE.
 - B.- THE SPARS, THE RIBS OR BULKHEADS AND THE STRINGERS OR STIFFENERS.
 - C.- THE METAL SPAR SHAPES, THE SKIN AND THE STRINGERS OR STIFFENERS.
 - D.- THE FLAPS, THE AILERON, THE SPEED BRAKES OR FLIGHT SPOILERS AND THE LEADING EDGES.
- 177.- WHICH IS THE AIRCRAFT PART THAT SUPPORTS THE AIRCRAFT DURING LANDING OR WHILE IT IS RESTING OR MOVING ABOUT ON THE GROUND? (19285) REF.: AC 65-15A, PAGE 23.
 - A.- THE FIVE MAIN PARTS OF THE AIRCRAFT.
 - B.- THE AIRFOIL AND THE POWER PLANTS.
 - C.- THE POWER PLANTS.
 - D.- THE LANDING GEAR.
- 178.- WHAT INSPECTION PROCESS CONSISTS IN MAGNETIZING THE PART AND THEN APPLYING FERROMAGNETIC PARTICLES TO THE SURFACE AREA TO BE INSPECTED? (18842) REF.: FAA-H-8083-30, PAGE 8-24.
 - A.- THE FERROMAGNETIC INSPECTION.
 - B.- THE LIQUID PENETRANT INSPECTION.
 - C.- THE ACOUSTIC EMISSION INSPECTION.
 - D.- THE MAGNETIC PARTICLE INSPECTION.

179.- WHAT ARE THE FUNDAMENTAL LAWS GOVERNING THE ACTION OF AIR ABOUT A WING? (19036) REF.: AC 65-15A, PAGE 30.

- A.- THE LAW OF VELOCITY AND SPEED.
- B.- THE BERNOULLI'S PRINCIPLE.
- C.- THE PASCAL'S LAW OF MOTION.
- D.- THE NEWTON'S LAW OF MOTION.

180.- WHAT MUST BE CAREFULLY DETERMINED IN ANY DAMAGED PART? (19125) REF.: AC 65-15A, PAGE 130.

- A.- THE ADJACENT PART.
- B.- THE FASTENER OF THE PART.
- C.- THE ALLOY OF THE PART.
- D.- THE FUNCTION OF THE PART.

181.- WHAT MUST BE KNOWN IN ORDER TO KNOW THE DENSITY OF A SUBSTANCE? (18654) REF.: FAA-H-8083-30, PAGE 3-2.

- A.- ITS VOLUME AND COMPOSITION.
- B.- ITS BASIC ELEMENTS AND ATOMIC WEIGHT.
- C.- ITS WEIGHT AND VOLUME.
- D.- ITS COMPOSITION AND GRAVITY.

182.- WHAT RIVETS ARE USED ON INTERIOR STRUCTURES? (18751) REF.: FAA-H-8083-30, PAGE 5-59.

- A.- THE SHEAR STRENGTH RIVET AND THE HEAT TREAT RIVET.
- B.- THE COUNTERSUNK HEAD RIVET AND THE SOLID RIVET.
- C.- THE FLATHEAD RIVET LIKE THE ROUNDHEAD RIVET.
- D.- THE BRAZIER HEAD RIVET AND THE UNIVERSAL HEAD RIVET.

183.- WHAT IS THE FIRST IMPORTANT CONSIDERATION IN THE HEAT TREATMENT OF A STEEL PART? (18737) REF.: FAA-H-8083-30, PAGE 5-19.

- A.- IS TO KNOW ITS PHYSICAL COMPOSITION.
- B.- IS TO KNOW ITS CHEMICAL COMPOSITION.
- C.- IS TO KNOW THE OVEN TEMPERATURE.
- D.- IS TO KNOW THE COOLING PROCEDURE.

184.- WHAT IS THE TERMINATING COMPONENT OF THE BASIC ENGINE? (19304) REF.: AC 65-12A, PAGE 59.

- A.- THE JET NOZZLE.
- B.- THE TAILPIPE.
- C.- THE EXHAUST CONE ASSEMBLY.
- D.- THE THRUST REVERSE.

185.- IN WHAT CONSISTS THE ENVELOPE METHOD OF COVERING WING WITH FABRIC? (18850) REF.: AC 65-15A, PAGE 93.

- A.- CONSIST IN GLUEING FABRIC OVER THE METAL TO MAKE AN ENVELOPE OR FOLD.
- B.- CONSIST IN EMBROIDERING COTTON TO MAKE AN ENVELOPE OR SLEEVE.
- C.- CONSISTS IN GLUEING FABRIC TO MAKE AN ENVELOPE OR SLEEVE.
- D.- CONSISTS IN SEWING FABRIC TO MAKE AN ENVELOPE OR SLEEVE.

186.- WHAT DOES THE EMPTY WEIGHT OF AN AIRCRAFT INCLUDE? (18717) REF.: FAA-H-8083-30, PAGE 4-4.

- A.- INCLUDES ALL OPERATING EQUIPMENT THAT HAS A FIXED LOCATION AND IS ACTUALLY INSTALLED IN THE AIRCRAFT.
- B.- INCLUDES ALL EQUIPMENT WEIGHT THAT HAS A MOVABLE LOCATION AND IS ACTUALLY CHECKED.
- C.- INCLUDES SOME OPERATING EQUIPMENT THAT HAS A POSITION AND ITS AIRWORTHINESS.
- D.- INCLUDES THE ENTIRE OPERATING EQUIPMENT THAT HAS A FIXED OR MOVABLE LOCATION AND IS ACTUALLY INSTALLED IN THE AIRCRAFT.

187.- WHERE IS LOCATED THE DISCHARGE NOZZLE IN A CARBURATOR? (19109) REF.: AC 65-12A, PAGE 115.

- A.- DIRECTLY IN THE MANIFOLD PRESSURE.
- B.- IN THE HIGH PRESSURE SIDE OF THE CARBURATOR.
- C.- IN THE THROAT PRESSURE SIDE OF THE VENTURI.
- D.- IN THE THROAT OF THE VENTURI.

188.- IN WHAT NUMERALS IS THE DIAL FACE OF THE TYPICAL ALTIMETER GRADUATED? (19274) REF.: AC 65-15A, PAGE 476.

- A.- IS GRADUATED WITH NUMERALS FROM ZERO TO NINE INCLUSIVE.
- B.- IS GRADUATED WITH NUMERALS, LETTERS AND SYMBOLS INCLUSIVE.
- C.- THE GRADUATION DEPENDS ON THE AIRPLANE MANUFACTURER.
- D.- THE GRADUATION DEPENDS ON THE INSTRUMENT MANUFACTURER.

189.- WHAT IS THE DENSITY OF A SUBSTANCE? (18653) REF.: FAA-H-8083-30, PAGE 3-2.

- A.- IS ITS VOLUME PER UNIT OF WEIGHT.
- B.- IS ITS WEIGHT PER UNIT OF VOLUME.
- C.- IS ITS COLOR AND WEIGHT PER UNIT OF VOLUME.
- D.- IS ITS CHEMISTRY COMPOSITION AND PHYSICAL RESISTANCE.

190.- IN HOW MANY SECTIONS IS THE DATA SHEET DIVIDED? (18817) REF.: FAA-H-8083-30, PAGE 8-5.

- A.- INTO ONE OR MORE SECTIONS.
- B.- INTO THREE OR MORE SECTIONS.
- C.- AT LEAST INTO TWELVE SECTIONS.
- D.- INTO FOUR OR SIX SECTIONS.

191.- WHAT TEMPERATURE WILL INDICATE THE CYLINDER HEAD TEMPERATURE INDICATOR PRIOR TO OPERATE THE ENGINE? (19307) REF.: AC 65-12A, PAGE 322.

- A.- WILL INDICATE THE CYLINDER TEMPERATURE.
- B.- WILL INDICATE THE FREE OUTSIDE AIR TEMPERATURE.
- C.- WILL INDICATE ANY FREE TEMPERATURE.
- D.- WILL INDICATE THE ENGINE GENERAL TEMPERATURE.

192.- WHAT IS THE CHORD OF AN AIRFOIL OR WING SECTION? (19040) REF.: AC 65-15A, PAGE 31.

- A.- IS A ROPE USED TO TIE THE WING FROM THE LEADING EDGE TO TRAILING EDGE.
- B.- IS A STRAIGHT LINE WHICH CROSSES THE WING FROM THE ROOT TO THE WING TIP.
- C.- IS AN IMAGINARY STRAIGHT LINE WHICH PASSES THROUGH THE SECTION FROM THE LEADING EDGE TO TRAILING EDGE.
- D.- IS A REAL STRAIGHT LINE WHICH UNITES THE SECTION FROM THE LEADING EDGE TO TRAILING EDGE.

193.- WHAT IS THE CENTER OF GRAVITY RANGE FOR AN AIRCRAFT? (18724) REF.: FAA-H-8083-30, PAGE 4-17.

- A.- IS THE AFT AND REAR BALANCE LIMITS.
- B.- IS THE LIMITS WITHIN WHICH THE AIRCRAFT MUST BALANCE.
- C.- ARE ALL THE DISTANCES THAT ARE CONSIDERED DURING WEIGHING.
- D.- IS WHERE THE WEIGH IS CONCENTRATED.

194.- HOW MAY BE CONSIDERED THE CENTER OF GRAVITY IN AN AIRCRAFT? (19287) REF.: AC 65-15A, PAGE 33.

- A.- AS A POINT AT WHICH THE BALANCE OF THE AIRCRAFT IS OPTIMUM.
- B.- AS A POINT AT WHICH ALL THE WEIGHT OF THE AIRCRAFT IS CONCENTRATED.
- C.- AS THE REAR AND FRONT POINT AT WHICH ALL THE WEIGHT OF THE AIRCRAFT IS LIMITED.
- D.- AS A PLACE WHERE ALL THE LOAD IN THE AIRCRAFT IS CONCENTRATED.

195.- WHAT DOES THE CARBURETOR HAVE IN ORDER TO SHUT OFF THE FUEL TO STOP THE ENGINE? (19107) REF.: AC 65-12A, PAGE 115.

- A.- A FUEL INYECTION AND A MANUAL START SYSTEM.
- B.- AN AUTOMATIC SELECTOR VALVE IN THE CARBURATOR.
- C.- AN IDLE CUTOFF SYSTEM.
- D.- A FUEL CONTROL IN THE COCKPIT.

196.- WHAT DOES THE CARBURETOR AIR TEMPERATURE GAGE INDICATE? (19312) REF.: AC 65-12A PAGE 431.

- A.- INDICATE THE CYLINDERS TEMPERATURE AND THE AIR AROUND THE CARBURATOR.
- B.- THE QUANTITY OF THE AIR THAT IS IN THE CYLINDERS.
- C.- THE TEMPERATURE OF THE AIR BEFORE IT ENTERS THE CARBURATOR.
- D.- THE PRESSURE AND THE TEMPERATURE OF THE AIR AFTER IT ENTERS THE CARBURATOR.

197.- WHAT DOES THE BLADE FLAPPING ACTION CREATE IN A HELICOPTER? (19023) REF.: AC 65-15A, PAGE 53.

- A.- CREATES AN UNBALANCE CONDITION WITH RESULTING VIBRATION.
- B.- CREATES A BIG LIFT CONDITION WITH RESULTING IN A SMOOTH FLIGHT.
- C.- CREATES SOME ADDITIONAL DRAG AND THE ENGINE HAS TO DEVELOP MORE POWER.
- D.- CREATES A BALANCE CONDITION WITH RESULTING IN LESS VIBRATION.

198.- WHAT IS USED TO DESIGN THE BASIC COMPONENT OF A CABLE? (18765) REF.: FAA-H-8083-30, PAGE 5-75.

- A.- NUMBERS AND LETTERS.
- B.- THE WIRE METAL.
- C.- THE AMOUNT OF WIRES IN EACH STRAND AND THE QUANTITY OF CABLES.
- D.- THE NUMBER OF STRANDS AND THE NUMBER OF WIRES IN EACH STRAND.

199.- WHICH IS USUALLY THE MOST CONVENIENT LOCATION FOR TEMPORARY BALLAST? (18644) REF.: FAA-H-8083-30, PAGE 4-23.

- A.- THE BAGGAGE COMPARTMENT.
- B.- THE CABIN COMPARTMENT.
- C.- THE TAIL COMPARTMENT.
- D.- THE WING ROOT COMPARTMENT.

200.- WHAT IS THE PURPOSE OF DEVELOPING SPECIAL TOOLS AND DEVICES? (19046) REF.: AC 65-15A, PAGE 133.

- A.- TO HELP THE MECHANIC MAKE HIS HOME-WORK ON TIME, COMPLEX AND UP TO DATE.
- B.- TO HELP THE MECHANIC MAKE HIS JOB SWIFT, THE BEST AND GOOD.
- C.- TO HELP THE MECHANIC MAKE HIS WORK SOFT, RELAXED AND QUICKLY.
- D.- TO HELP THE MECHANIC MAKE HIS WORK FASTER, SIMPLER AND BETTER.

201.- HOW MANY BRAKES APPLICATIONS CAN BE MADE WITH THE AIR BOTTLE THAT HAS NORMAL COMPRESSED AIR? (19268) REF.: AC 65-15A, PAGE 336.

- A.- SEVEN APPLICATION OF THE BRAKES.
- B.- ONE APPLICATION UNTIL A COMPLETE AIRCRAFT STOP.
- C.- SEVERAL APPLICATIONS OF THE BRAKES.
- D.- TWO FULL BRAKES APPLICATION.

202.- WHAT ARE THE ADJUSTABLE WRENCHS? (18858) REF.: FAA-H-8083-30, PAGE 9-5.

- A.- THEY ALSO ARE OPEN-END WRENCHES.
- B.- THEY CAN REPLACE HANDY UTILITY TOOLS.
- C.- THEY ARE HANDY UTILITY TOOLS.
- D.- THEY ARE MANUAL UTILITY TOOLS.

203.- WHAT DOES THE TENSION FORCE TRY TO DO? (18668) REF.: FAA-H-8083-30, PAGE 3-14.

- A.- IT TRIES TO COMPRES AN OBJECT.
- B.- IT TRIES TO CRUSH AN OBJECT.
- C.- IT TRIES TO PRESS AN OBJECT.
- D.- IT TRIES TO PULL AN OBJECT APART.

204.- WHICH IS A DOMINANT FACTOR THAT AFFECTS THE PHYSICAL PROPERTIES OF FLUIDS? (18697) REF.: FAA-H-8083-30, PAGE 3-23.

- A.- THE TEMPERATURE.
- B.- THE PRESSURE.
- C.- THE DENSITY.
- D.- THE ALTITUDE.

205.- HOW MAY THE STRESSED SKIN PANELS BE? (18825) REF.: AC 65-15A. PAGE 24.

- A.- THEY MAY BE EITHER SMOOTH OR BEADED.
- B.- THEY MAY BE STRESSED RESISTANCE OR LITTLE SMOOTH.
- C.- THEY MAY BE ROUND OR SQUARE.
- D.- ALL THEM ARE OVAL.

206.- WHY DOES STRESS CORROSION OCCUR? (18775) REF.: FAA-H-8083-30, PAGE 6-6.

- A.- STRESS CORROSION NEVER OCCURS.
- B.- STRESS CORROSION OCCURS AS THE RESULT OF MAINTAINING THE AIRPLANE DIRTY AND IN A DRY ENVIRONMENT.
- C.- STRESS CORROSION OCCURS AS THE RESULT OF KEEPING THE AIRPLANE OUT OF THE HANGAR.
- D.- STRESS CORROSION OCCURS AS THE RESULT OF THE COMBINED EFFECT OF SUSTAINED TENSILE STRESSES AND CORROSIVE ENVIRONMENT.

207.- WHICH MAY BE THE PRINCIPAL REQUIREMENT IN CERTAIN STRUCTURES? (19275) REF.: AC 65-15A, PAGE 1.

- A.- THE DURABILITY.
- B.- THE STRENGTH.
- C.- THE AIRWORTHINESS.
- D.- THE STRESSES.

- 208.- WHAT PROPELLER BLADES HAVE MORE RESISTANCE TO ABRASION? (18796) REF.: FAA-H-8083-30, PAGE 6-23.
 - A.- THE STEEL PROPELLER BLADES.
 - B.- THE ALUMINUM ALLOY PROPELLER BLADES.
 - C.- THE PURE ALUMINUM PROPELLER BLADES.
 - D.- THE MAGNESIUM ALUMINUM ALLOY PROPELLER BLADES.
- 209.- WHAT HAPPENS WITH SOME TYPES OF EXTINGUISHING AGENTS? (19294) REF.: AC 65-15A, PAGE 427.
 - A.- RAPIDLY CORRODE ALUMINUM ALLOY AND OTHER METAL.
 - B.- SLOWLY CORRODE ALUMINUM ALLOY AND OTHER METAL.
 - C.- ARE HARDENER TO ALUMINUM ALLOY AND OTHER METAL.
 - D.- LEAVE THE ALUMINUM ALLOY VERY BRIGHT.
- 210.- WHAT VERY LIGHT COLOR MAY APPEAR IN SOME STAINLESS STEEL ALLOYS METAL TEST? (18732) REF.: FAA-H-8083-30, PAGE 5-5.
 - A.- PINK.
 - B.- WHITE.
 - C.- BLACK.
 - D.- BROWN.
- 211.- SOME AIRCRAFT ARE REQUIRED TO BE WEIGHED AND HAVE THEIR CENTER OF GRAVITY CALCULATED ON A PERIODIC BASIS, TYPICALLY HOW OFTEN IS THIS? (18713) REF.: FAA-H-8083-30, PAGE 4-2.
 - A.- EVERY FIVE YEARS.
 - B.- EVERY THREE YEARS.
 - C.- ALMOST EVERY MONTH.
 - D.- WHEN THE OWNER WANTS.

212.- IF SOME AIRCRAFT ARE NOT WEIGHED WITH THE WHEELS ON THE SCALES, WHERE ARE THEY WEIGHED? (18723) REF.: FAA-H-8083-30. PAGE 4-16.

- A.- AT THE JACKING POINTS OR AT SPECIAL WEIGHED POINTS.
- B.- AT THE WHEELS POINTS OR AT SPECIAL WEIGHED POINTS.
- C.- AT THE JACKING POINT OR AT SPECIAL WEIGHED POINT.
- D.- AT THE NOSE AND TAIL POINTS OR AT SPECIFIC WEIGHED POINTS.

213.- WHICH IS THE PROPERTY OF SILICONE RUBBERS? (18745) FAA-H-8083-30, PAGE 5-35.

- A.- HAVE EXCELLENT HEAT STABILITY.
- B.- REACTS FAVORABLY WITH AROMATIC GASOLINES.
- C.- IS VERY EASE TO WORK.
- D.- RESISTS THE STRIKES VERY WELL.

214.- WHICH ARE THE MOST WIDELY USED SEMICONDUCTOR MATERIALS? (18976) REF.: FAA-H-8083-30, PAGE 10-3.

- A.- GLASS, CERAMIC AND PLASTICS.
- B.- SILICON AND GERMANIUM.
- C.- GOLD, COPPER AND SILVER.
- D.- MATERIAL WITH FREE ELECTRONS.

215.- IN WHAT FORMS AND SHAPES ARE STEEL AND STEEL ALLOYS MANUFACTURED? (18730) REF.: FAA-H-8083-30, PAGE 5-2.

- A.- ROLLER BARS, WELDING SHEETS, FLIXIBLE TUBING, FORGINGS AND CASTINGS.
- B.- THIN, THICK, MEDIUM, SMALL SIZE AND BIG SIZE.
- C.- BARS, SHEETS, TUBING, EXTRUSIONS, FORGINGS AND CASTINGS.
- D.- COOL, WARM, HOT, ICED, FROZEN AND REFRIGERATE.

216.- WHICH ARE THE INCREMENTS OF RIVETS SHANK DIAMETERS? (18760) REF.: FAA-H-8083-30, PAGE 5-64.

- A.- 1/54-INCH.
- B.- 1/16-INCH.
- C.- 1/32-INCH.
- D.- 1/64-INCH.

217.- WHAT ARE PERMITED TO REPAIR FABRIC-COVERED SURFACES? (18852) REF.: AC 65-15A, PAGE 99.

- A.- ONLY SEWN REPAIRS ARE PERMITED.
- B.- SEWN AND UNSEWN REPAIRS ARE PERMITED.
- C.- ONLY UNSEWN REPAIRS ARE PERMITED.
- D.- THE REPAIR DEPEND THE LONG DAMAGE.

218.- WHAT OTHER USE CAN BE GIVEN TO SEVERAL MATERIALS USED FOR THINNING, SPECIFIC PAINTS AND LACQUERS? (19122) REF.: AC 65-15A, PAGE 113.

- A.- THEY ARE ALSO AVAILABLE FOR SOLVENT CLEANING.
- B.- THEY ARE ALSO AVAILABLE FOR PAINT REMOVER AND CORROSION CONTROL.
- C.- THEY ARE ALSO AVAILABLE FOR PAINT FINISHING.
- D.- THEY ARE ALSO AVAILABLE TO JOIN COMPOSITES.

219.- WHAT PROCEDURES MUST BE OBSERVED WHEN SELECTING A SAW BLADE? (18861) REF.: FAA-H-8083-30, PAGE 9-9.

- A.- CHECK THE MATERIAL THAT YOU HAVE TO CUT.
- B.- SELECT AN APPROPRIATE SAW BLADE.
- C.- CHECK THE WEATHER.
- D.- CHECK THE ELECTRICAL GENERATOR.

220.- WHAT IS THE OBJECT OF SEALS IN THE AIRPLANES HYDRAULIC SYSTEM? (18746) REF.: FAA-H-8083-30, PAGE 5-36.

- A.- TO PREVENT FLUID FROM BEING CONTAMINATED.
- B.- TO PREVENT FLUID FROM PASSING A CERTAIN POINT.
- C.- TO RE-USE THE LIQUIDS MANY TIMES.
- D.- TO AVOID LOSS OF LIQUID AND TO CANCEL THE FLIGHT.

221.- WHAT ARE THE MOST COMMONLY USED THREADED FASTENING DEVICES ON AIRCRAFT? (18763) REF.: FAA-H-8083-30, PAGE 5-70.

- A.- THE BOLTS.
- B.- THE SCREWS.
- C.- THE FORKS.
- D.- THE STUD BOLTS.

222.- WHAT OPERATION OF MODERN AIRCRAFT IS DEPENDENT UPON THE USE OF INSTRUMENTS? (18804) REF.: AC 65-15A, PAGE 469.

- A.- THE FLIGHT OPERATIONAL SAFETY, ECONOMICAL ARRIVAL AND RELIABLE GROUND OPERATION.
- B.- THE ON-TIME DISPATCH, ECONOMICAL AND RELIABLE OPERATION.
- C.- THE MAINTENANCE WORK, FLIGHT SAFETY AND RELIABLE OPERATION.
- D.- THE SAFE, ECONOMICAL AND RELIABLE OPERATION.

223.- WHERE IS THE EXCEPTION IN THE USE OF ROUNDHEAD RIVETS IN THE INTERIOR OF THE AIRCRAFT? (18749) REF.: FAA-H-8083-30, PAGE 5-59.

- A.- WHERE CLEARANCE IS REQUIRED FOR ADJACENT MEMBERS.
- B.- WHERE EXTRA STRENGTH IS REQUIRED.
- C.- WHERE TOLERANCE IS REQUIRED FOR MAIN MEMBERS.
- D.- WHERE PASSENGER SEATS ARE CLOSE TO THE WINDOW.

224.- WHAT IS A TYPICAL ALUMINUM CORROSION TREATMENT? (18786) REF.: FAA-H-8083-30, PAGE 6-14.

- A.- REMOVE HYDRAULIC, SKIDROL, OIL AND SURFACE DIRT FROM THE ALUMINUM SURFACE USING ANY CAUSTIC SUITABLE MILD CLEANER.
- B.- REMOVE OLD PAINT, CORROSION, OIL AND SURFACE WASTE FROM THE ALUMINUM SURFACE USING ANY SUITABLE MILD CLEANER.
- C.- REMOVE OIL AND SURFACE DIRT FROM THE ALUMINUM SURFACE USING ANY SUITABLE MILD CLEANER.
- D.- REMOVE OIL, GREASE AND SURFACE DIRT FROM THE ALUMINUM SURFACE USING ANY SUITABLE MILD CLEANER OR SOAP.

225.- WHY ARE RELIEF VALVES USED IN PNEUMATIC SYSTEMS? (19266) REF.: AC 65-15A, PAGE 334.

- A.- TO PREVENT DAMAGE.
- B.- TO PROTECT THEE O-RINGS.
- C.- TO RELIEF PRESSURE.
- D.- TO MAINTAIN PRESSURE IN LIMITS.

226.- WHEN IS RELATIVELY LITTLE CORROSION TROUBLE EXPERIENCED WITH MAGNESIUM SKINS? (18783) REF.: FAA-H-8083-30, PAGE 6-9.

- A.- WHEN THE CORROSION IS ABSENT FROM THE SURFACE AND INSULATION IS ADEQUATELY MAINTAINED.
- B.- WHEN THE SURFACE FINISH HAS ZINC CHROMATE AND INSULATION IS ADEQUATELY COVERED.
- C.- WHEN THE REWORKED SURFACE FINISH AND INSULATION ARE ADEQUATELY MAINTAINED.
- D.- WHEN THE ORIGINAL SURFACE FINISH AND INSULATION ARE ADEQUATELY MAINTAINED.

227.- WHICH IS THE USE OF REAMERS? (20002) REF.: FAA-H-8083-30, PAGE 9-13.

- A.- THEY ARE USED TO SMOOTH AND ENLARGE HOLES TO EXACT SIZE.
- B.- THEY ARE USED TO ENLARGE HOLES TO EXACT SIZE.
- C.- THEY ARE USED TO POLISH METAL BEFORE PAINTING IT.
- D.- THEY ARE USED TO FIND CORROSION AND TO TREAT IT.

228.- WHAT ARE THE ANCIENT ENEMIES OF AIR TRANSPORTATION? (19134) REF.: AC 65-15A, PAGE 285.

- A.- THE FOG. CLOUD AND LIGHTNING.
- B.- THE METAL COST AND PROTECTION.
- C.- THE FUEL PRICE AND THE NOISE.
- D.- THE RAIN, SNOW AND ICE.

229.- WHAT DOES PROPELLER EFICIENCY DEPEND ON? (20001) REF.: AC 65-12A, PAGE 325.

- A.- HOW MUCH THE PROPELLER SLIPS.
- B.- THE CHORD LINE OF PROPELLER.
- C.- THE PROPELLER LENGTH.
- D.- THE NUMBERS OF BLADES.

230.- HOW MUCH VARIES THE PROPELLER EFFICIENCY? (19309) REF.: AC 65-12A, PAGE 325.

- A.- IT DEPENDS OF ENGINE POWER.
- B.- FROM 25% TO 50%.
- C.- FROM 75% TO 99%.
- D.- FROM 50% TO 87%.

231.- WHAT TYPE OF DEFECT WILL THE PENETRANT INSPECTION DETECT? (18821) REF.: FAA-H-8083-30, PAGE 8-19.

- A.- WILL DETECT SURFACE CRACKS OR POROSITY DEFECT.
- B.- WILL DETECT UNDER-SURFACE CRACKS OR VOIDS DEFECT.
- C.- WILL DETECT ELECTRICAL DISCONTINUITY OR FLAWS.
- D.- WILL DETECT INVISIBLE CRACKS OR SPOT POINTS.

232.- WHAT DOES AN ORIGINAL SURFACE TREAMENT FOR STEEL PARTS ALSO INCLUDE TO REMOVE ALL TRACES OF DIRT, OIL, GREASE, OXIDES, AND MOISTURE? (18790) REF.: FAA-H-8083-30, PAGE 6-17.

- A.- ALSO INCLUDES A WEAR RESISTANCE PROCEDURE.
- B.- IT INCLUDES AN ORIGINAL SURFACE TREATMENT.
- C.- USUALLY INCLUDES A CLEANING TREATMENT.
- D.- ALSO INCLUDES A PARTS REMOVAL.

- 233.- WHAT TYPE OF FABRIC OR CLOTH ARE USED IN THE MANUFACTURE OF AIRCRAFT COVERING? (18845) REF.: AC 65-15A. PAGE 85.
 - A.- DOMESTIC OR IMPORTED FIBERS.
 - B.- NATURAL AND ARTIFICIAL FIBERS.
 - C.- ORGANIC AND SYNTHETIC FIBERS.
 - D.- GLUED OR SEWING FIBERS.
- 234.- WHAT NONSTRUCTURAL MEMBERS OF AIRCRAFT CAN BE WELDED SATISFACTORILY? (19133) REF.: AC 65-15A, PAGE 259.
 - A.- ALL METALLICS MEMBERS.
 - B.- ONLY IRON ALLOY MEMBERS.
 - C.- ONLY STAINLESS STEEL MEMBERS.
 - D.- ONLY ALUMINUM ALLOY MEMBERS.
- 235.- ON LARGER AIPLANES, FROM PRIVATE BUSINESS JETS TO LARGE JUMBO JETS, IN RELATION OF WHAT ARE IDENTIFIED THE CENTER OF GRAVITY AND ITS RANGE? (18727) REF.: FAA-H-8083-30, PAGE 4-31.
 - A.- IN RELATION TO THE CHORD LINE.
 - B.- IN RELATION TO THE ROOT WIDTH OF THE WING.
 - C.- IN RELATION TO THE LENGTH OF THE WING.
 - D.- IN RELATION TO THE WIDTH OF THE WING.
- 236.- WHERE IS STAMPED THE ALLOY DESIGNATION ON LARGE ALUMINUM ALLOY TUBING? (18802) REF.: FAA-H-8083-30, PAGE 7-2.
 - A.- IN THE INTERIOR.
 - B.- ON THE SURFACE.
 - C.- WITH A TAG.
 - D.- WITH A FORM 8130-3 ATTACHED.

237.- IN WHAT CONSISTS A RADIAL ENGINE? (19303) REF.: AC 65-12A, PAGE 5.

- A.- IN A ROW OR ROWS OF CYLINDERS ARRANGED RADIALLY ABOUT A CENTRAL CRANKSHAFT.
- B.- IN A ROW OR ROWS OF CYLINDERS ARRANGED RADIALLY ABOUT A CENTRAL CRANKCASE.
- C.- IN A ROW OR ROWS OF CYLINDERS ARRANGED OPPSOSITE ABOUT A CENTRAL CRANKCASE.
- D.- IN A LINE OR LINES OF CYLINDERS ARRANGED IN LINE ABOUT A CRANKCASE.

238.- WHICH IS ONE OF THE DIFFERENCES BETWEEN A HELICOPTER AND A FIXED-WING AIRCRAFT? (18978) REF.: AC 65-15A, PAGE 49.

- A.- IS THE TYPE OF FLIGHT.
- B.- IS THE MAIN AND COCKPIT CABIN.
- C.- IS THE MAIN SOURCE OF DRAG.
- D.- IS THE MAIN SOURCE OF LIFT.

239.- WHEN IS IT NECESSARY TO MAKE REVISON TO A DRAWING? (18650) REF.: FAA-H-8083-30, PAGE 2-6.

- A.- WHEN THERE IS A CHANGE OF YEAR.
- B.- WHEN THERE IS A CHANGE IN AIRWORTHINESS CONDITION.
- C.- WHEN CHANGES IN DIMENSIONS, DESIGN OR MATERIALS ARE MADE.
- D.- WHEN THERE IS A CHANGE IN AIRCRAFT CERTIFICATION.

240.- WHAT MUST YOU NEVER DO WHEN WEIGHING AN AIRCRAFT? (18722) REF.: FAA-H-8083-30, PAGE 4-16.

- A.- WEIGH IT WITH THE FUEL TANKS EMPTY.
- B.- WEIGH IT WITH THE OIL TANKS PARTIALLY FULL.
- C.- WEIGH IT WITH THE HYDRAULIC TANKS PARTIALLY FULL.
- D.- WEIGH IT WITH THE FUEL TANKS PARTIALLY FULL.

241.- WHAT COMPOUND DOES THE SYNTHETIC RUBBER CALLED NEOPRENE HAVE? (18812) REF.: FAA-H-8083-30, PAGE 7-17.

- A.- IT HAS A PHOSPHATE BASE.
- B.- IT HAS A SYNTHETIC BASE.
- C.- IT HAS A NATURAL BASE.
- D.- IT HAS AN ACETYLENE BASE.

242.- WHAT METAL OR METAL ALLOY ARE NEARLY ALL ELASTIC STOP NUTS MADE OF? (18748) REF.: FAA-H-8083-30, PAGE 5-48.

- A.- THEY ARE MADE OF ALUMINUM OR STEEL ALLOY.
- B.- THEY ARE MADE OF COPPER OR MAGNESIUM ALLOY.
- C.- THEY ARE MADE OF IRON OR ALUMINUM ALLOY.
- D.- THEY ARE MADE OF STEEL OR ALUMINUM ALLOY.

243.- WHICH IS THE USE OF NACELLES AND OR PODS? (19280) REF.: AC 65-15A, PAGE 13.

- A.- TO HOUSE THE AUXILIARY POWER UNIT.
- B.- TO HOUSE BAGGAGE.
- C.- TO HOUSE THE ENGINES.
- D.- TO HOUSE FUEL.

244.- WHAT PRODUCES THE MOVEMENT ABOUT THE VERTICAL AXIS? (18992) REF.: AC 65-15A, PAGE 55.

- A.- FLARE.
- B.- DRAG.
- C.- YAW.
- D.- PITCH.

245.- SELF-LOCKING NUTS ARE USED ON AIRCRAFT TO (18833) REF.: FAA-H-8083-30, PAGE 5-46

- A.- PROVIDE TIGHT CONNECTIONS WHICH WILL SHAKE LOOSE UNDER SEVERE VIBRATIONS.
- B.- PROVIDE POOR CONNECTIONS WHICH WILL NOT SHAKE LOOSE UNDER SEVERE VIBRATION.
- C.- PROVIDE TIGHT CONNECTIONS WHICH WILL SHAKE LOOSE UNDER LIGHT VIBRATIONS.
- D.- PROVIDE TIGHT CONNECTIONS WHICH WILL NOT SHAKE LOOSE UNDER SEVERE VIBRATION.

246.- SURFACE CORROSION APPEARS AS A GENERAL: (18772) REF.: FAA-H-8083-30, PAGE 6-4.

- A.- ROUGHENING, ETCHING OR PITTING OF THE SURFACE.
- B.- CONTAMINATION OF THE METAL
- C.- WHITENNING OF THE METAL
- D.- HEATING OF THE SURFACE.

247.- WHAT HAS TO DO A MECHANIC IN ORDER TO MAKE A RIVET HOLE OF THE CORRECT SIZE? (19292) REF.: AC 65-15A, PAGE 169.

- A.- AT THE END DRILL A HOLE SLIGHTLY UNDERSIZE.
- B.- FIRST PUT GREASE AT THE HOLE.
- C.- FIRST DRILL A HOLE SLIGHTLY UNDERSIZE.
- D.- FIRST DRILL A HOLE SLIGHTLY OVERSIZE.

248.- IN PLACE OF WHAT, THE SPECIAL FASTENERS THAT PRODUCE HIGH STRENGTH WITH LIGHT WEIGHT, CAN BE USED? (18759) REF.: FAA-H-8083-30, PAGE 5-66.

- A.- IN PLACE OF STANDARD SA BOLTS AND NUTS.
- B.- IN PLACE OF CONVENTIONAL AN BOLTS AND NUTS.
- C.- IN PLACE OF STRENGTH AND LIGHT AN BOLTS AND NUTS.
- D.- IN PLACE OF ANY BOLTS AND NUTS IF THEY ARE THE SAME SIZE.

249.- WHAT AIRPLANE SECTION IS CALLED LEADING EDGE? (18822) REF.: AC 65-15A, PAGE 197.

- A.- THE FRONT SECTION OF WINGS, STABILIZERS OR OTHER AIRFOILS.
- B.- THE AFT SECTION OF WINGS, STABILIZERS OR OTHER AIRFOILS.
- C.- ANY SECTION OF WINGS, STABILIZERS OR AIRFOILS.
- D.- ANY STRUCTURAL PART OF AN AIRCRAFT.

250.- WHY AIR IS CONSIDERED A FLUID? (19286) REF.: AC 65-15A, PAGE 27.

- A.- BECAUSE IS A SUBSTANCE WHICH MAY BE MADE TO FLOW OR CHANGE IT'S SHAPE BY THE APPLICATION OF MODERATE PRESSURE.
- B.- BECAUSE IT HAS WEIGHT AND VOLUME.
- C.- BECAUSE IT CAN BE USED IN BALLOON AND TIRES.
- D.- AIR IS NOT CONSIDERED A FLUID.

251.- ABSOLUTE PRESSURE IS EQUAL TO REF.: FAA-H-8083-30, PAGE 3-25.

- A.- INSTRUMENT PRESSURE.
- B.- ATMOSPHERIC PRESSURE.
- C.- GAUGE PRESSURE PLUS ATMOSPHERIC PRESSURE.
- D.- GAUGE PRESSURE MINUS ATMOSPHERIC PRESSURE.

252.- ALL METALS AND ALLOYS ARE ELECTRICALLY ACTIVE AND HAVE REF.: FAA-H-8083-30, PAGE 6-3.

- A.- A SPECIFIC WEIGHT.
- B.- A SPECIFIC STRENGTH.
- C.- A SPECIFIC CORROSION RESISTANCE.
- D.- A SPECIFIC ELECTRICAL POTENTIAL.

253.- ARE THERE TIMES WHEN DEFINITE PRESSURE MUST BE APPLIED TO A NUT OR BOLT AS IT IS INSTALLED? REF.: FAA-H-8083-30, PAGE 9-7.

- A.- NO, IN THOSE CASES SPECIAL WRENCH MAY BE USED.
- B.- YES, IN THOSE CASES TORQUE WRENCH MAY BE USED.
- C.- NO, IN THOSE CASES COMMON WRENCH MUST BE USED.
- D.- YES, IN THOSE CASES TORQUE WRENCH MUST BE USED.

254.- AROUND WHAT IS AN AIRPLANE CONTROLLED IN FLIGHT? REF.: FAA-H-8083-30, PAGE 3-42.

- A.- IS CONTROLLED AROUND ONE OR MORE OF THREE AXES OF ROTATION.
- B.- IS CONTROLLED AROUND MAIN AND SECUNDARY FLIGHT CONTROLS.
- C.- IS CONTROLLED FROM THE COCKPIT.
- D.- IS CONTROLLED AROUND ONE AXIS OF ROTATION AT ONE TIME.

255.- AS A GENERAL RULE, WHAT WILL BE THE RIVET DIAMETER IN A REPAIR WORK? REF.: AC 65-15A, PAGE 165.

- A.- NOT LESS THAN THREE TIMES THE THICKNESS OF THE THINER SHEET.
- B.- NO LESS THAN THREE TIMES THE THICKNESS OF THE THICKER SHEET.
- C.- NO MORE THAN THREE TIMES THE THICKNESS OF THE THICKER SHEET.
- D.- EQUAL TO THREE TIMES THE THICKNESS OF THE THICKER SHEET.

256.- ASSUMING THAT THE TEMPERATURE AND PRESSURE REMAIN THE SAME, HOW DOES THE DENSITY OF THE AIR CHANGE WITH THE HUMIDITY? REF.: AC 65-15A, PAGE 29.

- A.- DEPENDS ON THE PROPORTION.
- B.- DOES NOT VARY.
- C.- VARIES INVERSELY.
- D.- VARIES DIRECTLY.

257.- AT LOW SPEED, WHAT ENGINES HAVE BETTER ECONOMY? REF.: AC 65-12A, PAGE 3.

- A.- THE RECIPROCATING AND TURBOPROPELLER ENGINES HAVE BETTER ECONOMY THAN THE TURBOJET ENGINES.
- B.- THE RECIPROCATING AND TURBOJET ENGINES HAVE BETTER ECONOMY THAN THE TURBOPROPELLER ENGINES.
- C.- THE TURBOPROPELLER ENGINES HAVE BETTER ECONOMY THAN THE TURBOJET AND RECIPROCATING ENGINES.
- D.- AT LOW SPEED ALL ENGINES BURN THE SAME QUANTITY OF FUEL.

258.- AT WHAT SPEED IS SAFER TO TAKEOFF AND LAND? REF.: FAA-H-8083-30, PAGE 4-1.

- A.- IS SAFER TO TAKEOFF AND LAND AT LOWER AIRSPEEDS THAN AT HIGHER SPEEDS.
- B.- IS SAFER TO TAKEOFF AND LAND AT HIGHER AIRSPEEDS THAN AT LOWER SPEEDS.
- C.- THE SAFEST TAKEOFF AND LANDING SPEED DEPEND ON PILOT'S ABILITY.
- D.- THE SAFER TAKEOFF AND LANDING SPEEDS DEPEND ON THE AIRPLANE MODEL.

259.- BY THE USE OF WHAT IS THE MASS OF AIR ACCELERATED WITHIN THE ENGINE? REF.: AC 65-12A, PAGE 65.

- A.- BY A CONTINUOUS-FLOW CYCLE.
- B.- BY A COMPRESSOR.
- C.- BY A TURBINE.
- D.- BY TWO AXLES.

260.- BASICALLY, WHAT ARE THE MAJOR SECTIONS OF A LARGE SINGLE-ROTOR HELICOPTER? REF.: AC 65-15A, PAGE 24.

- A.- THE ROTORS AND THE MAIN STRUCTURE.
- B.- THE TAIL CONE AND THE FUSELAGE.
- C.- THE CABIN AND THE TAIL CONE.
- D.- THE CABIN AND THE CARGO COMPARTMENT.

261.- BY WHAT MEANS METALS CAN BE JOINED? REF.: AC 65-15A, PAGE 247.

- A.- BY ANY MEANS LIKE BOLTING, RIVETING, WELDING, BRAZING, SOLDERING OR ADHESIVE BONDING.
- B.- BY AUTOMATIC MEANS LIKE BOLTING, RIVETING, SOLDERING MACHINE OR ADHESIVE BONDING MACHINE.
- C.- BY ELECTRICAL MEANS LIKE WELDING, BRAZING, SOLDERING OR ADHESIVE BONDING.
- D.- BY MECHANICAL MEANS LIKE BOLTING, RIVETING, WELDING, BRAZING, SOLDERING OR ADHESIVE BONDING.

262.- BY WHICH METHODS ARE NUTS, BOLTS, AND SCREWS SAFETIED WITH WIRE? REF.: FAA-H-8083-30, PAGE 5-80.

- A.- BY THE SHORT WIRE OR LONG AND EXTRA LONG WIRE METHOD.
- B.- BY THE SINGLE WIRE FOR ELECTRICAL CONNECTORS OR DOUBLE TWIST METHOD FOR PLUG AND BOLTS.
- C.- BY THE SINGLE WIRE OR DOUBLE TWIST METHOD.
- D.- BY THE TRIPLE WIRE OR UNIQUE TWIST METHOD.

263.- DURING HOVERING FLIGHT IN A NO-WIND CONDITION, HOW IS THE TIP-PATH PLANE? REF.: AC 65-15A, PAGE 50.

- A.- DEPEND ON THE PILOT SKILL.
- B.- HORIZONTAL, THIS IS VERTICAL TO THE GROUND.
- C.- VERTICAL, THIS IS PARALLEL TO THE GROUND.
- D.- HORIZONTAL, THIS IS PARALLEL TO THE GROUND.

264.- DURING VERTICAL FLIGHT IN A NO-WIND CONDITION, HOW DO LIFT AND THRUST FORCES ACT? REF.: AC 65-15A, PAGE 50.

- A.- VERTICALLY UPWARD.
- B.- VERTICALLY DOWNWARD.
- C.- HORIZONTALLY UPWARD.
- D.- HORIZONTALLY DOWNWARD.

265.- FROM WHAT METAL ARE REAMERS MADE? REF.: FAA-H-8083-30, PAGE 9-13.

- A.- THEY ARE MADE FROM EITHER VANADIUM TOOL STEEL OR HIGH-POWER STEEL.
- B.- THEY ARE MADE FROM EITHER CARBON TOOL STEEL OR HIGH-SPEED STEEL.
- C.- THEY ARE MADE FROM HARD TEMPERED STEEL.
- D.- THEY ARE MADE FROM EITHER CROMO NIQUEL VANADIUM STEEL OR LOW-SPEED STEEL.

266.- FROM WHAT ALLOYS ARE SOME WING PANELS FABRICATED? REF.: FAA-H-8083-30, PAGE 5-9.

- A.- FROM COPPER ALLOYS.
- B.- FROM IRON ALLOYS.
- C.- FROM MAGNESIUM ALLOYS.
- D.- FROM TITANIUM ALLOYS.

267.- HOW ARE AIRCRAFT DRAWINGS CONSIDERED? REF.: FAA-H-8083-30, PAGE 2-2.

- A.- EXPENSIVE AND VALUABLE.
- B.- IMPORTANT AND SECURE.
- C.- INVALUABLE AND CHEAPER.
- D.- GOODS AND HELPFUL.

268.- HOW ARE NEUMATIC SYSTEMS PROTECTED AGAINST DIRT? REF.: AC 65-15A, PAGE 335.

- A.- BY MEANS OF VARIOUS TYPES OF VALVES.
- B.- BY MEANS OF VARIOUS TYPES OF FILTERS.
- C.- BY MEANS OF VARIOUS TYPES OF PACKING AND O-RINGS.
- D.- BY MEANS OF VARIOUS TYPES OF LIQUIDS.

269.- HOW ARE MOST HELICOPTERS STRUCTURAL MEMBERS COMPARED WITH THOSE USED IN FIXED WING AIRCRAFT? REF.: AC 65-15A, PAGE 24.

- A.- ARE VERY DIFFERENT.
- B.- ARE THE SAME.
- C.- ARE SIMILAR.
- D.- ARE IDENTICAL.

270.- HOW ARE THE CONTROL SURFACES OF SOME AIRCRAFT COVERED? REF.: AC 65-15A, PAGE 18.

- A.- ARE COMPOSITE COVERED.
- B.- ARE FABRIC COVERED.
- C.- ARE FABRIC AND METAL COVERED.
- D.- ARE PLASTIC, FABRIC, METAL AND COMPOSITE COVERED.

271.- HOW ARE THE METHODS OF REPAIRING STRUCTURAL PORTIONS OF AN AIRCRAFT? REF.: AC 65-15A, PAGE 127.

- A.- THE METHODS ARE SPECIFIC BY EACH AIRCRAFT MODEL.
- B.- THE METHODS ARE NUMEROUS AND VARIED.
- C.- THE METHODS DEPEND THE CMA CATEGORY.
- D.- THE METHODS ARE SPECIFIC FOR EACH COUNTRY.

272.- HOW ARE USED IN EVERYDAY CONVERSATION THE WORDS SPEED AND VELOCITY? REF.: FAA-H-8083-30, PAGE 3-16.

- A.- SPEED LIKE QUICKLY AND VELOCITY LIKE FAST.
- B.- LIKE SYNONYM.
- C.- AS IF THEY MEAN THE SAME THING.
- D.- AS THEY WERE DIFFERENT THING.

273.- HOW ARE USUALLY ATTACHED THE NONSTRUCTURAL MEMBERS THAT ARE NOT REMOVABLE FROM THE HELICOPTER? REF.: AC 65-15A, PAGE 25.

- A.- THEY ARE ATTACHED BY BOLTING OR STRUCTURAL WELDING.
- B.- THEY ARE ATTACHED BY RIVETING OR SPOT WELDING.
- C.- THEY ARE ATTACHED BY GLUEING OR SPOT WELDING.
- D.- THEY ARE ATTACHED BY SOLDERING, BOLTING, RIVETING OR ANY KIND OF WELDING.

274.- HOW CAN AIRCRAFT FINISH BE CLASSIFIED? REF.: AC 65-15A, PAGE 113.

- A.- DECORATIVE, PROTECTIVE AND PRESERVER.
- B.- APPEARANCE, PRIMER AND PAINT.
- C.- PROTECTIVE, ANTI-DETERIORATION AND FINISH.
- D.- PROTECTIVE, APPEARANCE AND DECORATIVE.

275.- HOW CAN COMPOSITES MATERIAL BE MADE? REF.: FAA-H-8083-30, PAGE 5-34.

- A.- LIKE A SANDWICH STRUCTURE.
- B.- LAMINATED OR PLAIN.
- C.- WITH OR WITHOUT AN INNER CORE OF MATERIAL.
- D.- WITH OR WITHOUT AN EXTERNAL CORE OF MATERIAL

276.- HOW CAN MAGNETO IGNITION SYSTEMS BE CLASSIFIED? REF.: AC 65-12A, PAGE 177.

- A.- PRIMARY AND SECONDARY.
- B.- HIGH CURRENT AND LOW CURRENT.
- C.- HIGH IGNITION AND LOW IGNITION.
- D.- HIGH TENSION AND LOW TENSION.

277.- HOW DO CHANGES TAKE PLACE WITH SUPERSONIC FLOW IN VELOCITY, PRESSURE, TEMPERATURE, DENSITY AND FLOW DIRECTION? REF.: AC 65-15A, PAGE 59.

- A.- THE CHANGES TAKE PLACE SUDDENLY AND OVER A SHORT DISTANCE.
- B.- THE CHANGES TAKE PLACE SMOOTHLY AND OVER A SHORT DISTANCE.
- C.- THE CHANGES TAKE PLACE SMOOTHLY AND OVER A LONG DISTANCE.
- D.- THE CHANGES TAKE PLACE OVER ALL THE AIRFOIL.

278.- HOW DOES THE AIR FLOW OVER THE UPPER SURFACE OF AN AIRFOIL ACT, WHEN ITS SPEED OR VELOCITY INCREASES? REF.: AC 65-15A, PAGE 29.

- A.- THE PRESSURE INCREASES.
- B.- THE PRESSURE DECREASES.
- C.- THE AREA INCREASES.
- D.- THE VELOCITY AFFECTS THE PRESSURE.

279.- HOW DOES THE ENVIRONMENT AFFECT THE CONDITIONS UNDER WHICH AN AIRCRAFT IS MAINTAINED AND OPERATED? REF.: FAA-H-8083-30, PAGE 6-7.

- A.- THE ENVIRONMENT AFFECTS GREATLY THE CORROSION CHARACTERISTICS.
- B.- THE ENVIRONMENT AFFECTS ONLY MINIMALLY THE CORROSION CHARACTERISTICS.
- C.- THE ENVIRONMENT DOES NOT AFFECT THE CORROSION CHARACTERISTICS.
- D.- THE SALT WATER AND NOT THE ENVIRONMENT AFFECTS GREATLY THE CORROSION CHARACTERISTICS.

280.- HOW DO LIQUIDS AND GASES ACT AND HOW ARE BOTH CLASSIFIED? REF.: FAA-H-8083-30, PAGE 3-27.

- A.- BOTH ACT AS GASES AND ALSO ACT AS LIQUID.
- B.- BOTH ACT AS LIQUIDS AND ACT IN A VERY DISSIMILAR WAY.
- C.- BOTH ACT AS FLUIDS AND ACT IN A VERY DIFFERENT WAY.
- D.- BOTH ACT AS FLUIDS AND ACT IN A VERY SIMILAR WAY.

281.- HOW IS CALLED THE GEAR WITH THE IMPUT FORCE? REF.: FAA-H-8083-30, PAGE 3-11.

- A.- IS CALLED THE MAIN GEAR.
- B.- IS CALLED THE DRIVE GEAR.
- C.- IS CALLED THE DRIVEN GEAR.
- D.- IS CALLED THE FIRST CLASS GEAR.

282.- HOW DO MOST OF THE FORCES ACT ON THE FUSELAGE OF AN AIRCRAFT WHILE IT TAKES OFF? REF.: AC 65-15A, PAGE 131.

- A.- ACTS IN THE SAME DIRECTION.
- B.- ACTS IN THE OPPOSITE DIRECTION.
- C.- ACTS INDIVIDUALLY.
- D.- THE FORCES DO NOT AFFECT THE FUSELAGE.

283.- HOW DO STRESSES USUALLY ACT? REF.: AC 65-15A, PAGE 132.

- A.- THEY ACT BREAKING THE PIECES THAN DO NOT HAVE RIGHT TORQUE.
- B.- THEY ACT IN A HORIZONTAL MANNER.
- C.- THEY ACT IN COMBINATION RATHER THAN SINGLY.
- D.- THEY ACT SINGLY RATHER THAN IN COMBINATION.

284.- HOW FREQUENTLY ARE ALUMINUM ALLOY SHEETS MARKED? REF.: FAA-H-8083-30, PAGE 5-23.

- A.- EVERY FIVE INCHES OF MATERIAL.
- B.- EVERY SQUARE METER OF MATERIAL.
- C.- EVERY SQUARE FOOT OF MATERIAL.
- D.- EVERY SQUARE INCH OF MATERIAL.

285.- HOW HAS THE AIRCRAFT TO BE WEIGHED WITH RESPECT TO FUEL? REF.: FAA-H-8083-30, PAGE 4-16.

- A.- DEPENDS ON THE SCALE RANGE.
- B.- ONLY WITH EMPTY TANKS AND LINES.
- C.- ONLY WITH FULL FUEL IN THE TANKS AND LINES.
- D.- ONLY WITH RESIDUAL FUEL IN THE TANKS AND LINES.

- 286.- HOW IS CALLED THE AMOUNT OF FORCE ACTING ON A SPECIFIC AMOUNT OF SURFACE AREA? REF.: FAA-H-8083-30, PAGE 3-24.
 - A.- POWER.
 - **B.- PRESSURE**
 - C.- FORCE.
 - D.- QUANTITY.
- 287.- HOW IS CALLED THE MOVEMENT ABOUT THE LONGITUDINAL AXIS? REF.: AC 65-15A, PAGE 55.
 - A.- MOVEMENT.
 - B.- YAW.
 - C.- PITCH.
 - D.- ROLL.
- 288.- HOW IS CALLED THE BASIC BODY AND TAIL BOOM SECTIONS OF A TYPICAL HELICOPTER? REF.: AC 65-15A, PAGE 24.
 - A.- MONOCOQUE.
 - B.- REINFORCED.
 - C.- STANDARD.
 - D.- CONVENTIONAL.
- 289.- HOW IS CALLED THE SMALLEST PARTICLE OF MATTER THAT CAN EXIST AND STILL RETAIN ITS IDENTITY? REF.: FAA-H-8083-30, PAGE 10-1.
 - A.- THE ATOM.
 - B.- THE MOLECULE.
 - C.- THE ELECTRON.
 - D.- THE PROTON.
- 290.- HOW IS EACH TYPE OF RIVETS IDENTIFIED? REF.: FAA-H-8083-30, PAGE 5-59.
 - A.- BY ITS COLOR.
 - B.- BY A PART NUMBER.
 - C.- BY ITS FORM.
 - D.- BY ITS HEAD.

291.- HOW IS CALLED THE TENDENCY OF AN OBJECT TO REMAIN STATIONARY WHEN SUPPORTED FROM ITS OWN CENTER OF GRAVITY? REF.: AC 65-15A. PAGE 82.

- A.- DYNAMIC BALANCE.
- B.- STATIC BALANCE.
- C.- TRIM TABS BALANCE.
- D.- BALANCED CONDITION.

292.- HOW IS DEFINED MOTION? REF.: FAA-H-8083-30, PAGE 3-16.

- A.- IS A CONTINUED CHANGE OF POSITION OR PLACE.
- B.- IS CONTINUED CHANGE OF MASS AND DIAMETER.
- C.- IS AN ALTERATION IN POSITION OR PLACE.
- D.- IS SPEED OR VELOCITY.

293.- HOW IS POSSIBLE TO OBTAIN A MOMENT IN WEIGHT AND BALANCE? REF.: FAA-H-8083-30, PAGE 4-3.

- A.- DIVIDING THE WEIGHT BY ITS DISTANCE.
- B.- MULTIPLING THE WEIGHT BY ITS DISTANCE.
- C.- WEIGHING ALL THE ITEMS.
- D.- WEIGHING THE ENTIRE AIRPLANE.

294.- HOW IS POTENTIAL ENERGY DEFINED? REF.: FAA-H-8083-30, PAGE 3-3.

- A.- ENERGY THAT IS POWERFUL.
- B.- ENERGY THAT HAS HIGH POWER.
- C.- ENERGY THAT IS IN MOTION OR ENERGY THAT IS READY TO BE USED.
- D.- ENERGY THAT IS AT REST OR ENERGY THAT IS STORED.

295.- HOW IS THE IMPELLER WITHIN THE DIFFUSER CHAMBER LOCATED? REF.: AC 65-12A, PAGE 77.

- A.- IS LOCATED LIKE A DIFFUSER.
- B.- IS LOCATED PERPENDICULARLY.
- C.- IS LOCATED PARALLEL.
- D.- IS LOCATED CENTRALLY.

296.- HOW IS THE PROCEDURE FOR WELDING STAINLESS STEEL COMPARED WITH THE PROCEDURE FOR WELDING CARBON STEELS? REF.: AC 65-15A, PAGE 259.

- A.- IT IS EXACTLY THE SAME.
- B.- IT IS BASICALLY THE SAME.
- C.- IT COULD BE THE SAME.
- D.- BOTH PROCEDURE ARE ABSOLUTELY DIFFERENT.

297.- HOW IS THE STABILIZER USUALLY BUILT? REF.: AC 65-15A, PAGE 24.

- A.- LIKE AN AIRCRAFT FLAP, WITH SCREWS AND DEFLECTORS.
- B.- LIKE AN AIRCRAFT WING, WITH RIBS AND SPARS.
- C.- LIKE AN AIRCRAFT FUSELAGE, WITH LONGERONS AND RIBS.
- D.- IT IS BUILT LIKE A LANDING GEAR.

298.- HOW IS THE VELOCITY OF THE AIR AS IT FLOWS THROUGH THE VENTURI OF A CARBURATOR? REF.: AC 65-12A, PAGE 119.

- A.- THE VELOCITY INCREASES.
- B.- THE VELOCITY DECREASES.
- C.- THE PRESSURE INCREASES.
- D.- THE PRESSURE AFFECTS THE SPEED.

299.- DURING HOVERING FLIGHT IN A NO-WIND CONDITION, HOW IS THE TIP-PATH PLANE? REF.: AC 65-15A, PAGE 50.

- A.- DEPEND ON THE PILOT SKILL.
- B.- HORIZONTAL, THIS IS VERTICAL TO THE GROUND.
- C.- VERTICAL, THIS IS PARALLEL TO THE GROUND.
- D.- HORIZONTAL, THIS IS PARALLEL TO THE GROUND.

300.- DURING HOVERING FLIGHT IN A NO-WIND CONDITION, HOW IS THE TIP-PATH PLANE? REF.: AC 65-15A, PAGE 50.

- A.- DEPEND ON THE PILOT SKILL.
- B.- HORIZONTAL. THIS IS VERTICAL TO THE GROUND.
- C.- VERTICAL, THIS IS PARALLEL TO THE GROUND.
- D.- HORIZONTAL, THIS IS PARALLEL TO THE GROUND.